

August 11, 1958

# Aviation Week

*Including Space Technology*

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Methods Used  
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A McGraw-Hill Publication

Douglas DC-8 Jet Transport



**NASA Details Space Program Contracts**

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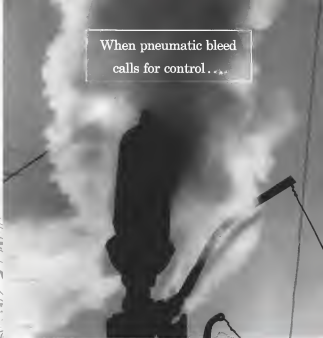
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### AVIATION CALENDAR

Aug. 15-15—Annual Western Regional Meeting, American Astronautical Society, Danforth, Anaheim, Stanford University, Palo Alto, Calif.  
Aug. 18-22—Joint General Meeting, American Institute of Electrical Engineers, Hotel Statler, Sacramento, Calif.  
Aug. 19-22—Western Electronic Show & Convention, Institute of Radio Engineers, Ambassador Hotel, Los Angeles, Calif.  
Aug. 23-27—Third Annual Convention, National Flying Club Assn., (Incorporated), Roseville Hotel, Hollywood, Calif.  
Aug. 19-19—Second Symposium on Naval Hydrodynamics, Washington D. C.  
Aug. 19-19—North America Congress, in Combined International Federation, Amsterdam, Holland.  
Aug. 25-31—1958 Trans and Southwestern Meeting, General, Grand Prairie, Tex.  
Aug. 19-22—1—Annual One Design Sailplane Regatta, Hana, Maui, Hawaii, N. T.  
Aug. 21-22—1—Auto Race, Professional Race, Fort Worth, Ind. (Purdue), Ohio (Cleveland), 15 Hudson Ave., Akron, Ohio.  
Sept. 1-7—1958 Fairbairn Flying Display and Exhibition, Society of British Aircraft Constructors, Fairbairn, Eng.  
Sept. 2-12—Symposium of High Powered Radar Group, Research Program, Massachusetts Institute of Technology, Cambridge, Mass. (Security clearance required).  
Sept. 1-10—1958 Symposium, Engineering Conference, Massachusetts Institute of Technology, Cambridge, Mass.  
Sept. 4-6—NAAU Annual Food Meeting, Cambridge University, Cambridge, Eng. land.  
Sept. 4-15—First International Congress of the Astronautical Sciences, Prince Hotel, Madrid, Spain.  
Sept. 9-13—Second National Conference on Applied Meteorology, Engineering Assn. (Continued on page 6)

### AVIATION WEEK Including Space Technology

August 11, 1958

Vol. 48, No. 6

Designed specifically for the aviation engineer, this issue contains the latest news and developments in the field of aviation. It includes a special section on space technology, featuring the latest news and developments in the field of space exploration. The issue also contains a comprehensive guide to the latest products and services available in the aviation industry. This is a must-read for anyone involved in the aviation industry.

Subscription: \$5.00 per year, \$2.50 in advance. Single copy: \$1.00. Payment in advance. Please send no money back.

AVIATION WEEK, August 11, 1958

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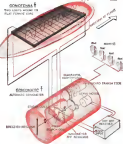
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## EDITORIAL

# AVIATION'S ECONOMIC PROBLEMS

The eagerly awaited Chermignon report (see page 28) which was distributed to Congress and appropriate federal agencies last week provides a sharp focus on the economic problems facing the aviation industry and a keen analysis of that industry's growing role as a significant force in the national economy. The Chermignon report is the first comprehensive look at the role of civil aviation in the national economy since the Peifer Committee report a decade ago.

The intervening decade has been marked by major economic and technical growth trends that have not only altered the role of aviation in the overall economy. Dr. Paul W. Chermignon of Harvard University, who prepared the report, and Elwood R. "Pete" Quisenberry, the President's special assistant on aviation policy, who accompanied Dr. Chermignon to do the job, both deserve the industry's commendation for the manner in which this job was done.

### Possible Benefits

There is an opportunity due the report to serve two useful purposes.

• To serve as a lubricant to ease friction between some segments of the industry and some agencies of the government which has been as effective deterrent to constructive progress. The report makes it abundantly clear that the economic problems now facing the industry are not the result of imprudent planning or action by any particular group but rather result from a combination of interlocking circumstances for which almost everybody in industry and government must shoulder some responsibility.

• To serve as a catalyst in stimulating a coordinated plan of action aimed at solving these problems by all the government agencies involved and both the air transport and nonairtransport portions of the industry. The report does not offer any ready-made answers for all of the economic problems looming on the near horizon, but it does suggest several broad avenues down which a well coordinated industry-government program could move successfully.

There are already indications that action is stirring in some areas cited in the report. The complex administration of the Military Air Transport Service and the non-military airlines is already being reevaluated at the Department of Defense level and the Air Coordinating Committee is exploring avenues of financing the large scale sale of jet-powered transports that are a necessary auxiliary to the U.S. airlines shift to gas turbine power plants.

### Where Initiative Lies

There is no question that much of the initiative in moving toward solution of these economic problems must come from the federal regulatory and financial agencies involved in the many facets of civil aviation.

But it would be a tragic mistake by the aviation industry to lean heavily on government action alone. Government action such as that of the transportation tax, liberal government financing policies on second hand equipment for the export market, new look at the airline fare structure by the Civil Aeronautics Board and a larger share of MATS non-military traffic to commercial airlines can only provide the industry with more capital resources and more operational latitude to solve its own problems.

The major aspects toward greater and lasting solutions to these economic problems must come from within the industry itself. It will require extremely vigorous, intelligent and thoughtful leadership by the management of each individual airline and manufacturers involved to permit growth through the financial problems of the jet age.

Air transport has become so vital and integral a part of the American economy that it would be unthinkable to have it collapse from economic stress. The life raft of government subsidy is still needed to prevent this. But it would be a tragic misstep if the airline industry, which in recent years has been almost completely off federal subsidy and which a dramatic posture can inflict on the nation's economy, were again to fall back on this device and again become a direct drain on the taxpayers.

### Top-Level Attention

It should be a source of great encouragement to the aviation industry in its efforts to achieve long-term solutions to the current crop of economic problems to note the consistent and vigorous attention that has been devoted to this area by the highest levels of the government during the past few years. Beginning with the Harding report, which laid bare the impending crisis in air traffic control, continuing with the monumental Curtis report that spawned the new Federal Aviation Agency and now punctuated by the Chermignon report on economic problems of the jet age, the executive branch of the government has developed an effective working approach to the most dynamic aspect of our expanding economy. It has recruited able men to tackle these jobs and, with solid congressional support, has translated their recommendations into action. As we pointed out last week, the record of solid achievement during the current session of Congress has been truly remarkable.

With the new Federal Aviation Agency ready to conceive new air traffic and the prospect of gaining vigorous leadership for it from Mr. Quisenberry as its initial head and with James E. Hill, current Civil Aeronautics Administrator, in his depth and essential capacity, the outlook for civil aviation is brightening despite the still-termed accurate thunder showers on the immediate horizon.

—Robert Holt



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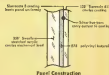
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## WHO'S WHERE

### In the Front Office

Mark M. Townsend, president, Rentr Airline, Inc.

Madeline B. Ladden, president, Langley Corp., San Diego, Calif., succeeding L. M. Ladden who remains in local charge.

A. C. Danks, vice president in charge of development engineering, General Fanucor Corp., Santa Ana, Calif.

The Industrial Products Group of Minneapolis-St. Paul, Minn., has acquired the following subsidiaries:

Werner & Hoyer, vice president, Pacific Seawater, Inc., Culver City, Calif.

Charles A. Hill, vice president-director, Acousticon Corp., Culver City, Calif., succeeding Tracy S. Clark, chief.

Sam F. Amy, vice president, Inc. Air Engineering Corp., Santa Monica, Calif.

J. Louis Frank, vice president-electrical, Olin Products, Inc., New York, N. Y.

Charles Theodore, vice president, Inc. Engineering, Inc., Culver City, Calif.

The Shulster Corp., subsidiary of Jewish Airline Corp., Dayton, Ohio, has announced the following appointments:

John E. Walsh, vice president field sales, William T. Walsh, vice president engineering, Thomas W. Clark, assistant vice president service facilities division.

Maj. Gen. Russell E. Wilson, Deputy Assistant Chief of Staff for Health-Med, USAF Headquarters, Washington, D. C.

Howard M. McGee, assistant to the commander, Westinghouse Aircraft Co., New York, Calif.

**Honors and Elections**

Dr. Marvin J. Kelly, president of Bell Telephone Laboratories, has been named 1957 recipient of the John F. Kennedy Award for his achievements in electronics, leadership, and contributions to the defense of the country.

Dr. Robert J. Johnson, president of Data Control Systems, Inc., has been appointed a member of the National Research Council to represent the National Science Foundation in the Division of Engineering and Industrial Research for a period of three years.

Dr. George S. Coombs, Professor Emeritus of Polytechnic Institute at the University of Pennsylvania, School of Engineering, has been elected to the National Academy of Sciences.

Merlin M. Dicker, president of the Decker Corp., has been elected vice president of the National Aeronautics Association of the United States, Washington, D. C.

(Continued on p. 17)

## INDUSTRY OBSERVER

■ Nite Zent will receive its first order for the Doppler radar developed by Ryan Aeronautical Co., which is either intended for use on major radar sites or for enabling Nite Zent to discriminate between fast-moving near and slow-moving targets.

■ Initial attempts already have been made to attach a lightweight Nocturnal Station Station, satellite and other satellites are about ready for launching.

■ When intercontinental ballistic missiles will have a shattering skin as opposed to the stainless steel surface of the Atlas.

■ Report to Defense Department on recommendations for research with hydrogen to promote its use in aircraft and missile structures is being submitted by a panel of National Academy of Sciences' Materials Advisory Board. Panel members include representatives from Boeing Corp. of America, East Bayfield Co., Miami, General Electric, Atomic Energy Commission and Nuclear Materials Inc.

■ Thrustor Metals Corp. of America is shipping its first batch of new test results for the T-1A1-3000 in North American Aviation and Chevrolet-Vauxhall. Report, and may be used for market at present in North American's B-70 Mach 3 bomber (AW July 18, p. 23).

■ De Havilland has developed a new version of the Gyron Junior jet engine designed in Mark 18. Although the engine is in its final work, de Havilland says the 7,000 lb. thrust of the earlier Mark 1 is "greatly enhanced."

■ Soviet Union is putting 10 times the effort into geological research, particularly discovery of natural deposits, in the U. S. is, according to information recently obtained by U. S. Geological Survey. Soviet effort, plus the increasing demand for solid fuel propellants and low and pressure resistant alloys and compounds generated by missiles and high speed aircraft, has prompted the United States Geological Survey to ask Congress for an increase of 35 million dollars in this country's current geological research program.

■ Tests at Wright Air Development Center have shown that a man in strict clothes can withstand temperatures of 150° for 30 min., while a man in flying gear can stay in 240° heat for 15 min.

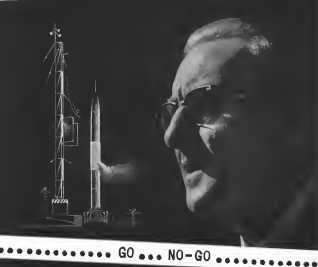
■ Convair's latest three-to-five-place L-34 helicopter under development reportedly will have a maximum speed of about 100 mph, and a ceiling of 14,355 ft. The aircraft, single rotor drive, is intended for passenger transportation, pilot training, small amphibious work and observation. It will carry a three-man crew.

■ Special Projects Office of Guy B. Farnes Engineers, New York City consulting engineers with heavy experience in underground installations, is conducting a study on tunnel, shaft design and construction for the Martin Marietta International Airfield. The study is being conducted under a contract from Air Research and Development Command's Ballistic Missile Division. Findings could result in "hard" landing sites that would be quite capable of withstanding even direct hits by megaton warheads (AW July 21, p. 15).

■ Failure of latest Vanguard satellite launching vehicle last June 25 occurred when Avco's second stage engine cut off prematurely due to low chamber pressure.

■ Initial small quantities of nitrogen tetraoxide ordered by North American's Rockwell International and Avco-General for evaluation are being delivered up with orders for 10-ton test lots. Long storage life makes the oxidizer a promising substitute for the considerably more powerful, but less stable, liquid oxygen.





**Automatic checkout equipment is a project of J. G. Ferguson, Senior Staff Engineer, Stavid Engineering, Inc.**

Mr. Ferguson has specialized in the development of frequency standards for Loran and other navigation systems, field transmitters measuring systems, radiofax, ray tube displays, and many other test and measuring devices for electronic equipment. His current work of Stavid is to develop equipment for reliability studies and automatic testing of electronic systems. Mr. Ferguson is one of a team of Stavid scientists and engineers who are applying their knowledge... from constant research production... to projects of major importance to the defense and progress of our country.

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## Washington Roundup

### Security Veil Tightened

While House, concerned with the Middle East situation and agitated by what it believes to be attempts of the individual agencies to reach goals their own ways in applying its policy of "no news is good news" with serious risks.

One service public information officer in the Pentagon complained last week that "Rumsfeld can give out more information than we can." Another said, "All we can handle here are news releases."

Information going beyond the routine release of press releases, the most pass through the office of William S. Gaudin, Assistant Secretary of Defense for Public Affairs and former aide to presidential press secretary James Hagerly, where it is graded for "policy" and "credibility" as well as for actual accuracy which also plays a part fixed in the last two. Additional rules power over the services was recently granted Searles, in a move to pursue the individual services from releasing information that might conceivably be interpreted as prejudging inter-service results, or of flouting the party line as laid down by the Administration.

### Federal Aviation Agency

Chances are good that the proposed bill to create a new Federal Aviation Agency will be approved by Congress and sent to the White House this week. Cardenas, a vice chairman late last week, to work out differences between the House and Senate-approved version. House amendments to the Senate bill were merely trimmed.

### Capital Gains Ruling

Airlines cannot keep capital gains from the sale of flight equipment without debiting the amounts from their taxable payments. Civil Aeronautics Board ruled last week. In a decision in the Capital Gains Ruling, CAB said owners it has no legal right to withhold such amounts but can legally use them to offset its annual long-term payments of interest on equipment with regulations allowing the method to use the gain for new equipment purchases.

### Attempt for Perspective

Top Pentagon officials who are going out of their way to try and avert any perception of the public criticism touched off last December by McNamara's fall, and shortly, change to launch a Vietnam strategy. Widespread criticism and disappointment at the Vietnam failure stressed generally from advance press billing that the staff study would attack the current the Secret Service's large Spentz II and II and years that the U.S. had not lost its industrial superiority.

Last week, in a press briefing on the Air Force team plan planned for later this month (AW Aug. 4, p. 35), Advanced Research Project Agency Director Roy W. Johnson carefully explained that there "is a slight possibility" of success and that "otherwise a lower odds is not unhelpful." Johnson, packed into speaking by articles detailing the team plan which Defense Department faced might re-create the pre-Vietnam atmosphere, added.

The complete success of these first experimental flights depend on the perfect functioning of more than

300,000 complex and interdependent parts, the failure of any one of which could nullify the complete success of the flight. Certainly, the chances of success are no better than one in 10."

Concomitantly, a brief Air Force statement confirming the first successful firing of an Atlas reconnaissance satellite made with all three engines ignited outboard against one thought that the launch meant the U.S. had an operational ICBM equivalent. The release quoted Air Force Secretary James H. Dugan as saying that "in the future the Atlas test program will be expected when flight testing such highly complex weapons systems."

### Restricted Flight Area

Strategic Air Command plans to conduct simulated bombing missions over the Great Lakes region has resulted in a Civil Aeronautics Administration decision last week denying all civilian flights to certain special CAA air traffic control approval for operations in the designated North Central Restricted Area during specified dates during August, September and October.

The military exercises will be conducted between the hours of 1 a.m. and 4 a.m. (CDT) at altitudes up to 40,000 ft. Areas involved is roughly bounded by Detroit, Springfield, Saginaw and Ironwood, Mich., in a rectangular area extending to the Canadian border. Effective dates of restrictions are: Aug. 23, 24, 25, 27, 28, Sept. 3, 4, 5, 10, 11, 12, 13, 16, 19, 20, Oct. 1, 2, 3, 14, 15, 16, 17. Areas will be used only one of three nights total per week in first two months and four nights per listed week in October. Air Route Traffic Control facilities will have information 13 hours in advance.

### Dallas to West Coast

Civil Aeronautics Board last week voted to consider, and scheduled portions of the Dallas to the West Service Case with the Southwest Transportation Service Case 7 and route change in Dallas to the West Coast route under which allows Continental Airlines and Texas Texas Air mass increased rotation. Board approval for nonstop service in the case (AW July 13, p. 10).

Scope of the consolidated proceeding will cover the need for single carrier service from the Florida cities of Miami, Orlando, Jacksonville and Tampa to St. Petersburg/Clearwater along with Atlanta and Birmingham to San Diego, Los Angeles and San Francisco/Oakland. The case also includes the need for such service connecting Houston with the three California cities. Inter-airline routes proposed on the routes involved are New Orleans, Houston, San Antonio, El Paso, Dallas, Ft. Worth, Albuquerque, Tucson, Phoenix and Las Vegas.

### ATA Appeal

Air Transport Association is asking the Civil Aeronautics Board to reconsider its order of July 15 denying the airline group permission to hold exclusive with joint fare discussions. ATA and the existing complaint of 11 airlines have discounts, which it termed a "70 new evaluation" of individual carrier tariff filings, has greatly hampered industry efforts to accelerate the sale of tickets. It added that more than 40 possible fare combinations are possible between New York and Miami.

—Washington staff

# NASA Details Plans for Space Challenge

Nation's new space agency asks for \$343 million to begin drive to cut Soviet technological lead.

By Fred Eastman

Washington—Detailed plans by which the U.S. hopes to challenge Soviet advances in space technology at an initial cost of \$543.1 million have been revealed to Congress by the newly created National Aeronautics and Space Administration.

To lay the groundwork for a long-range civilian space exploration program and continue or expand projects already under way, Congress has been asked to approve new appropriations of \$123 million for Fiscal 1976. The remainder of the \$543.1 million would be acquired through transfer of \$117 million from defense department accounts, some of it from the Advanced Research Projects Agency recently authorized civilian projects under \$101.1 million from the National Aeronautics and Space Administration, which is to be absorbed by NASA.

Total amount needed for Fiscal 1976 represents only a fraction of the funds that will be required in future years, Dr. Hugh D. Dryden, NASA's director, told Congress. When projects now in research and development reach the production stage the costs will skyrocket to many times the current hard request.

## Fund Division

Of the new money asked, \$7 million is for salaries and expenses, \$59.2 million for research and development and \$47.3 million for new construction and equipment. This, along with the money to be transferred, will give NASA a total of \$481.1 million for salaries and expenses, \$357.7 million for research and development and \$76.8 million for construction and equipment.

The \$7 million asked for salaries and expenses will provide for 800 additional positions in the new NASA, and above the \$480 million employed by NASA. Salaries of the new positions will range from about \$14,000 a year to the lower grades up to a maximum of \$33,000 annually in the top administrative, scientific and engineering grades.

The \$47.3 million asked for new construction includes \$24.5 million for expansion of NASA facilities at Wallops Island, Va., \$15.7 million for construction of a new space projects center at Beltsville, Md., and \$7.1 million for equipment and instrumentation at various locations.

The \$76.8 million additional requested for research and development will be used to broaden the current space program and add new projects to cover a wide field ranging from lunar and interplanetary probes to manned space flight and development of advanced thrust rocket engines. Two supplements to the \$117 million for programs either started or planned by ARPA which are civilian in nature.

## Construction Plans

Funds for construction have been requested in two separate appropriations, \$13 million to start on existing facilities of NASA or support of both agencies for construction. Under the new Space Act, however, appropriations requests exceeding \$300,000 must have prior authorization. To eliminate any confusion, Congress quickly authorized such requests.

Under the statute, NASA did not request prior congressional authorization for construction. Under the new Space Act, however, appropriations requests exceeding \$300,000 must have prior authorization. To eliminate any confusion, Congress quickly authorized such requests.

A breakdown of planned construction, expenses and instrumentation projects for NASA and NASA budget requests follows.

Wallops Island, Va.—NASA, \$130,000; NASA, \$24.5 million. Total—\$154.5 million. This is required to provide for the general launching facilities, land acquisition, construction, instrumentation and other facilities required for launching of solid and liquid fuel rockets.

Land acquisition and construction of essential facilities are estimated at \$6,381,500. Construction includes an administrative and support control building, model assembly and reliability data building, advanced systems shop, emergency and other supporting facilities.

The \$130,000 asked by NASA is for construction.

Island facilities will total \$14,546,000 and include launching pad, control room, \$55,000, launching site control station, \$546,000, vehicle and model launching facilities, \$51,122,500.

Beltsville, Md.—NASA, \$117 million. \$113.5 million in Wallops area. • Space vehicles, aerial and surface probes and tracking system, \$1,482,000.

- Long-range solar tracking system, \$2,500,000.
- Radar space guide and test equipment, \$750,000.
- Tracking data handling equipment, \$900,000. Total electronic tracking system, \$7,130,000.
- Satellite and missile cameras, \$150,000.
- Long focal-length tracking telescope, \$150,000. Total photographic and optical system, \$500,000.
- High-gain telemetry receiving antenna system, \$1,300,000.
- Standard telemetry receiving and recording systems, \$400,000.
- Special narrow-band, long-range telemetry system, \$300,000. Total telemetry data receiving and recording system, \$1,600,000.
- Launch site programming and control system, \$500,000.
- Range safety and destruction system, \$200,000.
- Computation system, \$600,000.
- Meteorological instrument devices, \$750,000. Total range safety, operations and control system, \$2,750,000.

## Space Projects Center

Space Projects Center at Beltsville, Md.—NASA, \$117 million. The construction of two buildings, a research projects laboratory and a space projects building along with supporting facilities, is proposed.

The space projects building, to be constructed at an estimated cost of \$1,020,000, will serve as the initial headquarters for the center in which offices for the director and his staff and the central staff and consulting staff will be located.

One wing in the basement will house computer equipment, the other wing a control room.

The projects laboratory will provide space for engineering personnel, electronics work and control and guidance instrumentation. The basement will house instrumentation and electronics development equipment. Cost is estimated at \$1,045,000.

Equipment and instrumentation at various locations, NASA, \$19,550,000.

One of the first tasks to be taken over by NASA is the general tracking of satellites. For this purpose, Dryden said the House Space Committee, NASA will begin by taking over the present network of the International Geophysical Year. In addition, NASA will take over the Vanguard Computing Center in Washington and add to that as well as make use of optical systems operated under contract by the

Southwest Institute and the other optical stations.

Also is a breakdown of the instrumentation system and the estimated costs.

- Improvement of existing Minuteman stations, \$4,100,000.
- Provision of equipment and facilities for two additional Minuteman stations, \$2,000,000.
- Improvement of capabilities of existing optical tracking system \$3,000,000.
- Provision of long-range scheduler data acquisition system, \$2,576,000. Total for extension of Minuteman and optical network capabilities, \$5,100,000.
- Position range and velocity measuring system, \$1,340,000.
- Telescope reception equipment, \$1,500,000.
- Data handling and transmission system, \$750,000. Total data handling and transmission system, \$4,400,000.
- Terminal data transmission equipment, \$250,000.
- Tracking system, \$1,515,000.
- Computing system, \$750,000.
- Command-control system, \$175,000.

## Space Technology

# NASA Drafts Broad Research Program

By Fred Clark

Washington—Nation's new civilian space agency plans to start one aspect of a million projects, thrust, solid rocket engine, Air Force projects of the Rover nuclear motor project and a new in space power at its most into a boat, \$117.7 million research and development program for fiscal 1976.

The research funds requested by the National Aeronautics and Space Administration for its first year would be \$115 million in salaries and materials, \$115 million in salaries and materials, \$115 million in salaries and materials, \$115 million in salaries and materials.

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Total research command-control system, \$1,000,000.

• Portable launching and instrumentation system, \$1,400,000.

Langley Aeronautical Laboratory, Hampton, Va., NASA budget, \$11,546,000.

This has been requested for the construction of a high temperature atomized dynamometer facility and a 110-kilowatt solid state between cost and weight.

Levitt Laboratory

Levitt Flight Propulsion Laboratory, Cleveland, Ohio, NASA, \$7,000,000. The program at Levitt includes installation of an engine in the engine test tunnel, modification of the air track tunnel rocket engine, testing propulsion engine laboratory tunnel, hypersonic ramjet propulsion facility, hypersonic ramjet propulsion facility, hypersonic ramjet propulsion facility, hypersonic ramjet propulsion facility.

Ames Aeronautical Laboratories, Moffett Field, Calif., NASA, \$1,720,000.

Included in this request are funds

for a 12 by 12 in. hypersonic helium tunnel, hypersonic ramjet laboratory and modification of the flight research laboratory.

Under the space act the administration has given a certain amount of money to supporting personnel and being subject to a limited number of its projects.

In its report, engineering and administrative personnel, the administration was authorized to appoint a maximum of 350 persons at a top level of \$79,000 annually, 10 of which could exceed that up to a \$120,000 a year level. In addition, the administration, if found it necessary, could extend the contract for its scientific and engineering personnel without previous governmental agreement at two grades higher than that provided under the Classification Act of 1949 to Grade 7 rather than Grade 5.

Under the organization plan submitted to Congress, NASA has asked for a total of 57 special positions which will be determined by the administrator at rates equal to or in excess of \$17,730 annual salary.

velopment. The engine is expected to produce 1.5 million lb thrust and is at least three to four years away.

• Hypersonic engine studies. That \$6.9 million budget item includes development contracts for a 12,000 lb thrust ramjet engine at a cost of \$2.1 million, an 80,000 lb thrust engine using the same technology at \$1.5 million, and a 100,000 lb thrust liquid oxygen engine, \$1 million. In addition, \$1.7 million was asked for design work and adaptation of a Thor or Jupiter intermediate range ballistic missile for engine flight tests.

• Launching of lunar and interplanetary probes and orbital probes and geophysical, astronomical and meteorological satellites from 700 lb to 2,000 lb, with vehicle weight from 10,000 and 20,000 to 40,000 and 50,000.

• Manned space capsules that could serve for the Mars atmospheric balloon probes to reach a stage of reliability that the capsule could be used for an orbit at some 120,000 miles.

A full-scale model of the capsule will be built as it already has undergone some ground tests, including all motion simulation, at NASA's Langley Laboratory. Scale models have been fired upstream in a hypersonic

## Budget Estimates

National Aeronautics and Space Administration Fiscal 1976 projected estimates for research and development (in millions of dollars):

Program	Contracted Work*				Supplies and Materials	Equipment	Total
	Gov.	Non-Gov.	Minerals	Cost			
Space Science	117.3	115.7	45.7	11.5	11.5	94.7	
Space Technology	12.6	46	17.3	4.5	4.5	16.4	
Meteorology and Communications							
Task	0.1	4.22	5.70	.48	11.5		
Technology of Manned Space							
Flight Vehicles	2.0	6.50	21.90	....	30.0		
Total	132.2	177.50	57.82	17.28	116.2	116.2	

\* Contracted research, development, operations, technical services, repairs, alterations and major construction.







### McDonnell Helicopter Shows Lifting Capacity

McDonnell Model 120 rotor helicopter is powered by three Allison QT455 gas turbine engines, compared to a diesel engine and a turboprop engine on the previous jet engine. Mounted on each of the three rotor blades tip, where fuel is injected and burning occurs. Rotor system, designed by McDonnell for the AV-14 Counterair (AV-14C, 24, 1955, p. 13), eliminates power shunting, gear loss, clutch and tail rotor. Helicopter, which has empty weight of 2,400 lb., is designed to lift one level lift at maximum gross weight with one gas turbine engine inoperative. Maximum weight is 4,000 lb. Long lead is point-to-point.

great deal of negative-type control over military research and development through the budgeting and expenditure processes.

"It said that 'it is quite possible' for a project to be subjected to an error in some budgetary process at the level of the Services of Defense.

"Final operations should be paid to guarantee research and development, and to support it. The committee, and that committee, points, 'are still suffering under the role of unannounced expenditures and other false functions.' The Defense Department was charged to define its various budgetary procedures and submit a report

Four-Atomic Energy Commission on the population into program.

Tests are conducted at the National Aeronautics Testing Station in Idaho and are subject to controlled testing. Experiments done last year at the Idaho facility. Purpose is to maintain same main safety of reactor operation by improving predictability of the behavior of fusion products when they are released in the atmosphere at low levels. Elaborate sampling network, through 3,200 yd. from the Idaho point, and it will be used to monitor the effects of various weather conditions on downward airborne radioactivity and ground contamination.

An Air Force Special Weapons Center of Air Research and Development Center is managing the experiment, and participants include the Atomic Energy Commission's Idaho Operations Office, Atomic Nuclear Population Department of General Electric and Atomic Energy Project of the University of Rochester.

### Scientists Report Space Flight Gains

San Diego—Technical advances indicating promising lines of attack on problems attendant to space flight were outlined in papers delivered here last week in sessions of Space Exploration Meeting sponsored jointly by Air Research and Development Command, Aerospace Rocket Society and Institute of Aeronautical Sciences.

Repeated meeting was attended by some 450 representatives, had a closed session on space research programs. The major session on air research and life, as well as technology. A panel on economic and social aspects of space exploration was conducted, with Theodore van Kernen III, A. J. Steward of Cal Tech, Mission Alpha of USAF Office of Scientific Research, Director of Advanced Research, D. N. Shickel, Dunlop & Associates, A. C. Hall, Martin Co. Diers or Division, all participating.

Research efforts in materials have special focus results which tend to show strong trends approaching those of atomic collisions may be possible, according to one paper, while another dealt with proper siting of population centers, type of gas accelerator to produce efficiency in space travel.

Details of development of Navy's geostationary of Naval Development Center also presented as was a comparison of interest in orbital research, orbital research, along with other studies, orbital and other launch factors aspect of space flight.

Communications, power supplies, inspection accuracy requirements were tested in a panel, with a status report given on attitude control for satellites.

## Beirut Tests One-Manager Airlift Concept

By Robert E. Farnell

Evros, France—A significant airlift job performed by USAF's 312nd Air Division during Lebanon crisis clearly demonstrated efficiency of a "single manager" system for operating an airlift task force.

The 312nd Air Division headquarters at Evros, carried out the bulk of the Air Force transport effort during the Lebanon "crisis period" July 15-27 without a hitch. Some 405 sorties were flown by 312nd assets which shuttled around the clock between Evros and New East ports.

Nearly eight million pounds of cargo were loaded during the 11-day period. More than 1,000 cases were carried, including 7,500 fully-equipped U.S. Army combat troops.

The 312nd's success also resulted from its ability to support Beirut port operations in that country.

Airlift task force commander for the operation was Col. Clyde Rex, 312nd Air Division Commander. Col. Rex reported directly to Gen. Frank F. Bivens, Commander in Chief USAF.

### C-130 Field Capability

The Lebanon airlift also pointed out the field capability of the Lockheed C-130 Hercules helicopter transport. 312nd Air Division crews had just completed humanitarian training, on the far side of the Lebanon job came up. Grounding pace of the airlift meant that most of the division's C-130s were there just about inspection periods, not on "active" breakdowns.

One of the Lebanon crew, according to 312nd's efforts, was Douglas C-130 Hercules transport. Both 312nd's C-130s and C-130s were the backbone of the airlift which was the division's Hercules C-130. The 312nd's Hercules lost a hand. The Hercules C-130 Hercules played a supporting role to the main lift effort.

In this, field maintenance has been shown by USAF on the side of the 312nd Air Division in the Lebanon effort. Throughout the airlift a top secret label was slipped on all 312nd operations. American Wings. In maintaining 312nd efforts and on even here at Evros, has proved highly the following account of how the USAF's Lebanon airlift was carried out between July 15-27.

Within hours after the July 14-15 evacuation began, all 312nd assets were put on the alert. Operations "Blue Bell," code name for the New East airlift, began July 15. The big NATO air base at Athens, Turkey, was alerted in the advance staging area for U.S. lead-



USMC signs at Beirut Airport terminal building porte Douglas C-124C Globemaster.

ing operations in the New East. While Sixth Fleet Marines were landing at Beirut with the evacuation on July 15, on the same day 312nd assets began to airlift U.S. Army paratroopers from Germany into Asia.

As Operations Beirut unfolded, Air Lift Task Force Commander Col. Rex had available 46 C-130 transports from the division's 517th Troop Carrier Wing stationed at Evros, 12 C-124 Globemaster transports from the division's 312nd Troop Carrier Squadron based at Rhein-Main, Germany, and about 30 C-119s from the division's 60th Troop Carrier Wing based at Dorn, France. These assets, plus 16 C-124s turned over to the 312nd by MATS Korea Air Transport Force,

formed the main airlift effort into the New East.

Flight of 41 C-119s from Beirut AFB, Texas, and Andrews AFB, Md., flew a tactical support mission from the United States directly to the Lebanon staging area, but their aircraft reports were immediately downgraded to the U.S. Air Force.

### March to Athens

Within the 14-day period during July 15-16, 312nd assets airlifted "Alpha Force" from Frankfurt-Hahn, West Germany, to Athens. The lift was carried out in 72 sorties by 34 C-130s, 16 C-124s and 12 C-119s. Alpha Force consisted of 1,200 paratroopers of the 24th Infantry Division, a unit of the U.S. 7th Army, plus 9,000 lb. of cargo to keep the paratroopers supplied for five days.

Aircraft carrying Alpha Force took off from Frankfurt-Hahn at 10:00 a.m. into the Beirut C-119 units flew over the Alps and crossed Greece to reach Athens. A flight of 1,400 lb. to the state. Almost immediately, however, conditions in Athens were relaxed by Sergeant Andrew and George. As a result 312nd C-130s were leaving their home base at Evros, flying to Frankfurt-Hahn, to pick up loads, then backtracking around the Alps to Munich, where they landed to school. Final leg was a tight corridor down the Mediterranean Sea to Athens. Then instead of 1,500 lb. the C-119 airlift only was scheduled as 2,500 lb. (Major C-119 and C-124 assets were routed through Capodakia Airport at Naples, Italy).

Alpha Force was to be transferred as quickly as possible to a forward port of arrival. But by July 17 it was apparent the Lebanon situation would remain harsh, quiet, so Alpha Force was

### Mikoyan Emphasizes Manned Aircraft Need

Moscow—Russian designer A. I. Mikoyan again with Red Air Force plans that manned aircraft will not be overwhelmed by unmanned missiles for long.

Mikoyan, interviewed by "Kommunisticheskaya Pravda," declared that the substantial difference in speed and altitude of light man-carrying between unmanned ballistic missiles and piloted airplanes is important. "In the future," he said, "these two types of flying apparatus will merge into one."

"Obviously the fast armed Sybair will be eventually a super-speed airplane based on the upper limits of the atmosphere as a winged missile."

Mikoyan predicted that neither planes nor missiles will be able to carry out a mission into the upper atmosphere for launching to great heights.





See, and intimate action which picks up passengers from an interstate flight.

#### Airline Fears

Scheduled carriers fear that they can be faced with unstarred competition from the intimate carrier on flights within Alaska. Pending CAB action from the Alaska preferential in the Southwest Case.

Priorities had asked authority for the hiring of additional pilots, plus regular schedules for passenger transportation. On the issue of a report by chairman Herbert K. Brown, the Board is now engaged in an advisory process, one which will allow these operators to be classified as Alaska or line operators controlled by an individual, person or a corporation. In addition, the emergency recommendations they permitted to land the pilots and operate without adequate facilities except where asked asked action have been taken. The proposed rule change would be limited to a two-year experiment.

Confidential sources also contend that a lack of authority over interstate action might open the door for intrastate routes to move in and tap the lucrative traffic of the interstate, leaving the state during the winter months.

It is possible that the new regulations could go into effect before the tentative advisory sessions scheduled and in

ample time for these or ten operators to manage their respective operations and later increase their route within a direct competition with the scheduled carriers without any federal endorsement except CAB safety rules.

A week before selection to the possible intra-state problem is the hope that Alaska will have a commission to control that segment of the industry.

One such proposal is for the location of an Alaska Civil Aeronautics Board under an Alaska Air Commission Act which would take effect upon completion of legislative action. As suggested by the Board, the Board would have five members with authority similar to that of the CAB. The Alaska Board appointed by the governor would have the right to question the setting of fares or adjustment of carrier under the CAB's administrative authority.

CAB power over carrier now operating between Alaska and the U.S. would be in a two-year experiment. The Board would be empowered to set rates on fares for routes as opposed to the present interstate and commerce ranges.

Some observers feel a change of the structure of the Alaska Civil Aeronautics Board would be a good idea. A session control over it provides that Congress has the power to continue during laws until the new state takes action of its own.

## Airlines to Buy Doppler on Gamble

Washington—Major intrastate airlines, several domestic carriers and three aviation manufacturers intend to gamble with their money to buy and build 8,000 new Doppler radar on a scheduled basis as made of most Federal Communications Commission carriers (AW, Aug. 1, p. 27).

Although the Commission denied an airline petition to create its order action to move civil Doppler radar out of 8,000 new lines to 12,000 new lines, FCC indicated its willingness to grant regular licenses, normally for five years, for 8,000 new airline radar and to attempt to provide reasonable construction period for such equipment.

Despite qualifications and edge cases, FCC's new rule, which will allow national and several domestic carriers appear ready to gamble on a reasonable license for the \$380 million. Offshore the Air Line, Radio, Radio, Radio, Radio and General Electric, plus the United States, are in the process of design and manufacture of 8,000 new equipment. Radio Corp. of America, which had requested its 8,000 new Doppler radar, is now in the process of design and manufacture of 8,000 new equipment.

Airmetrical Radio, Inc., and Air Transport Air will neither recognize nor discourage airlines from proceeding with 8,000 new, but will only require to make individual evaluation.

## Three Carriers Win New Southern Steps

Washington—Eastern, National and Southern airlines last week awarded tentative CAB Aeronautics Board approval for new routes in an expansion of the South East Local Service Case (AW April 21, p. 37).

CAB order would allow Southern to serve Houston and Augusta, Ala., in addition to flight Atlanta, Fla., between New Orleans and Miami, Fla.

Eastern would serve Melbourne, Fla., on its Miami-Melbourne route.

National also would serve Melbourne on its Florida-Melbourne route, along with competitive authority to provide one daily nonstop to Orlando on the airline's exchange service with Delta and American between Miami and points west of Tulsa.

## Vickers Explains Viscount Controls

The following communication has been received from Vickers Armstrong Aircraft Ltd., Weybridge, England, in reference to the "Viscount Controls" article by John S. Hooton, former editor, Radio Aviation, Ca., Chula Vista, Calif., and published on page 133 of *Airline World's* Aug. 4 issue.

"With reference to the letter concerning Viscount controls published in your last issue, we regret to state that, although the Manchester accident investigation itself concerned only airline locks, a complete investigation was naturally made by Vickers of all the other controls lock systems.

"It is completely impossible for any state sheet or even for installed flight to operate these locks. Major structural failure would be necessary first. Nevertheless, further attempts have been made to strengthen the Board's such things have not been considered in a practical sense to the integrity of the British industry and all the relevant government agencies. The suggestion that the Board should have controls around the point of the locks which is to prevent getting, etc., leaving the control system when the aircraft is on the ground. If the locks are in the cockpit part of the system, they are the past controls are left right along the controls in the cockpit being re-acted.

Major overhaul, therefore, applied which cannot be statistically evaluated for fatigue calculations. That is, a complete overhaul of the system of operation of the locks. The suggestion of leaving the locks in the cockpit position that requiring tension on the cable for locking is a definition is a full description of the controls in the cockpit, and that the controls are interconnected so that to lock one is to lock both.

"The last two paragraphs of the letter are further explanations which are the official control of the aircraft. They had the benefit of top engineering and scientific evidence from Farnborough, A.R.S., B.E.A. and Vickers. This evidence included results of most careful tests and flight tests.

"The writer of this letter, in fact, admits, admits he has not seen the evidence or even a full report of the system and it is difficult to understand it. He has his own view, those of highly qualified and independent engineers of world repute who spent months on the case.

"It is a fact, furthermore, that the Viscount is a safety record which has been well known in the industry for a number of years, and that over 173 of these aircraft are now in daily service all over the world with a reputation as all airports which is second to none."

## U. S. Carriers Gird for Bilateral Battle

By L. L. Doty

Washington—France's denunciation of its bilateral air pact with the U.S. last month (AW July 26, p. 34) is affecting domestic airlines' attitudes against one another more greatly to face the fight against within the U.S.

Most U.S. airlines, particularly the transcontinental carriers, look upon the French move as a tactical maneuver designed to pressure the State Department into a more generous attitude toward the French. These carriers are worried that during the next 12 months the period of greatest competition will actually commence—the State Department will concede to at least some of France's demands.

Both international and domestic airlines will fight any action that will increase competition. But the concern of domestic airlines is more deep-seated since they already have seen a significant amount of traffic diverted from domestic routes to the newly established routes between the West Coast and Europe.

France is definitely making a policy move between the West Coast and Paris to match those made by TWA and Pan American. During the final months of the latest bilateral negotiations between France and the U.S., a policy move was offered Air France by the State Department, but it was rejected.

France wants a far broader concept of a route exchange between the two countries than is now contained in the bilateral agreement. In effect, the intention is to allow Air France to operate to some extent in U.S. gateway cities in America now served by U.S. international traffic opening into Paris. This move Air France wants to duplicate the TWA domestic routes.

The policy move request was enough to create protests from transcontinental and domestic airlines. But any move to meet Air France's status into domestic routes would mean a loss of revenue.

Always, their attitude has declared that the point of traffic privileges between San Francisco and New York being sought indirectly by Qantas Airways (AW May 15, p. 11) would set a dangerous precedent. And, they add, any guarantee established by surrendering to Air France's demands would open the floodgates of U.S. traffic to all foreign carriers.

The State Department is giving no clue as to its position on the move except to say that the U.S. is ready to engage negotiations whenever France is ready. One State Department official was of opinion to make it clear that the

renunciation of the bilateral deal did not put a dam on air service between France and the U.S.

He went one step further and announced that the State Department would not stop one step after the bilateral was terminated, explaining that termination of the agreement did not automatically stop foreign carrier permits. He did admit that, after the 12 month grace period, the two countries could "agree to disagree" and drop the matter but evidenced hope that sufficient would be reached before then. He confirmed that the State Department feels Air France has "a good thing."

"In any negotiating on routes, France has two points against her from the start. In the first place, approximately 50% of all Air France's international traffic is U.S. traffic. Secondly, the carrier has nothing to offer the U.S. in additional traffic in exchange for those it seeks.

#### Threat Debated

Whether Air France will be willing to surrender to have a share of its total traffic by carrying out the threat to terminate the bilateral agreement remains to be seen. Some observers believe that France is serious in its warning and that it feels U.S. carriers will be unwilling to begin to negotiate a treaty in its major cities, even though the percentage of traffic taken to all U.S. international traffic is only 2%.

Previously, some observers felt that France is betting that the two U.S. airlines involved, Pan American and TWA, would refuse to accept Frenchly granted permits in the event of a breakdown in the agreement and that this difference will leave the State Department no alternative than to concede to Air France's demands.

On the point of route exchange, France contends that reciprocity was the intent of the 1946 agreement. At the time of the agreement, neither France nor the United States had international air service that would permit point-to-point and other nonstop operations.

The agreement was made American gird carriers the right to operate from New York in the U.S. to Paris and vice versa. It allows Air France to operate from anywhere in France to Boston, New York, Chicago and Houston.

Domestic airlines agree that this is fair exchange of traffic with Air France being the better of the deal. The French government agrees that it is unreasonable to prohibit Air France from competing with U.S. carriers in the U.S. while the latter can compete with Air France in France.

One of the most on the beginning table is that France has nothing to offer the U.S. that U.S. carriers wish want.

The French threat to withdraw from the bilateral agreement has been almost a year ago when the Civil Aeronautics Board granted TWA and Pan American the right to fly the polar route from the West Coast to London and Paris. Early this year, the French indicated the threat to drop the pact as negotiations, which resumed last October, began to flag (AW March 3, p. E2).

The French government explains that there are no political considerations in this move. It is a business move and that the action was based on commercial reasons only. Although to join the denunciation on the new de-Casle government is discredited by the State Department on grounds that the move began long before Gen. Charles de Gaulle came to power.

The international community under norms the strike routes throughout the world and the movement of the movement of their strikes for national prestige and as a means of strengthening national economy. This factor is opposed to the operating policy of smaller airlines as well as the larger carriers.

For example, unaffected behind Air France's fight for U.S. traffic privileges, is a bid by the privately-owned French carrier, TAF, to bid for contracts in major U.S. cities through the West Coast via Honolulu. Although the French government is not pressing this request at present, it is not, by any means, being shelved.

France's officials actually expect to negotiate in large a share of the lucrative Australian tourist market in their own and are seeking a wider range of U.S. gateways as a means of obtaining that market. However, some believe that the move is merely a tactical move since competition within the U.S. is already so keen that load factors generally are trending downward and that the U.S. airlines are in a tight spot.

According to the Air Transport Association, 75% of all traffic between the U.S. and other countries was carried on U.S. airlines in 1949, and the foreign airlines were carrying the U.S. market. By 1957, the number of foreign carriers operating into U.S. markets had grown to 39, and only 63% of the total traffic flow and the U.S. flag.

Behind the move is a growing fear among both international and domestic carriers that the Soviet Union's Aeroflot international airline may eventually gain a strong foothold in U.S. markets and that the airlines that have already been established in bilateral with other countries.



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## SHORTLINES

► **Alaska Airlines** has begun a new larceny service between Portland and Seattle and Fairbanks and Anchorage using custom built Douglas DC-8C aircraft. The airline is scheduling daily weekday flights, except Saturdays, from Seattle to Fairbanks with an extra over-night evening flight scheduled from Seattle to Fairbanks on Friday. Departures are scheduled from Portland Tuesday, Thursday and Friday.

► **Bozell International Airways** reports a 44.7% increase in its air transport sales for the first five months of 1973—\$86,068 versus last year's comparable 474,135 express ton miles for the same period of 1972. Bozell flew 2,718,675 freight ton miles in the first five months of the year, a 26.5% increase against 2,160,673 freight ton miles in the first five months of 1972.

► **British European Co.** has sold a British Aerospace BAe 147 to Air Charter Ltd., an independent British operator, for use as a troop carrier and general charter operations. The aircraft will be fitted to transport more than 100 passengers or fully equipped troops. Given for 23 transport in a special 133 tonne troop and cargo version are also being studied for British Ministry of Supply and RAF Transport Command.

► **Emery Air Freight Corp.** reports a second quarter 1973 net income of \$87,318, a 2.2% increase over last quarter net income of \$81,930.

► **International Air Transport Assn.** has awarded IATA's American National Airlines a transatlantic charter, as IATA's 40th member company.

► **Midwest Airlines** is replacing its eight flight agents with female stewardesses on all 81 flights. From 1947 through 1969, Midwest employed male hostesses. As of Jan. 1, 1970, the transition to male flight agents was completed. Then, on July 1, 1970, the airline again began using hostesses. Transition will be completed by next April 30.

► **Northeast Airlines** operating 300-ers for the first six months of 1973 totaled \$44,156,306, an increase of 15% over the same period of last year. Net income for the first half of 1973 was \$521,213, including \$429,479 obtained by property disposal. Operating for 1972 began totaled \$608,945, made possible by income of \$1,895,032 from disposal of flight equipment. The airline's total June 1973 operating revenues were \$9,968,273, up 39% from June 1972.

## AIRLINE OBSERVER

► **Battle between flight engineers' and pilots' unions** has become intensified as a result of conflicting reports by both groups' leaders (AWN Aug. 4, p. 41 and AWN July 28, p. 12). Both unions are determined to stand firm in their opposition to one another just for the present at least, in a series of strikes against carriers introducing jet transports that year and early next year may be in the offing. Flight Engineers' International Assn. stressed significant last week with Eastern Air Lines after a ringing denunciation of pilot-related last finding board which earlier recommended that the airline require that crew members be pilot-qualified.

► **Watch for Canadian government** to quash any competitive threats against Trans-Canada Airlines on the airline's transatlantic monopoly in a result of a recent report by a British aviation consultant who suggested only token competition on domestic intensity routes for TCA for some time in the future. Heavy traffic routes between Montreal, Toronto, Winnipeg and Vancouver are expected to be protected for the government-owned airline. Other carriers, many multi- and two-engine aircraft, will continue under the present strict regulations. Report also says that operations of single engine aircraft and helicopters be allowed freedom to operate whenever they can find business... a suggestion opposed by most Canadian aviation organizations who feel such competition may result in a price war.

► **English firm of International Aerotec Ltd.** will supply ground ILS equipment to Hungarian civil aviation authorities and aircraft ILS units to MARJAV, the Hungarian airline. Contract agreement calls for installation of PTC 1000 ground ILS equipment at Budapest, the Budapest airport, in Hungary. Aircraft will be equipped with STC units for which a limited number of licenses have been issued following a delay in negotiations held up by an expense shortage on the airborne equipment.

► **Nine level of Lockheed Electra** will be "substantially lower" than that of present four engine propeller aircraft, according to recent study conducted by the Port of New York Authority. The Port Authority said it expects no problems in handling the landing phase which is quite a 147 ft. from start of takeoff, then present aircraft are at a point 22 ft. above. Tests were made at indicated air speeds from 130 to 180 kt. to simulate typical airline conditions.

► **Nine level of the Soviet Tu-104** has found it to use Le Bourget terminal instead of Orly Field on the Paris-Moscow route. Paris Airport Authority reports the Aeroflot jet was delayed to the less densely populated area because of the "excessive noise of the jet's engines."

► **Japan Air Lines** given to more as additional L-401,200 shares of stock. The shares, with a face value of nearly \$2 million, will increase JAL capital from \$199 million to \$219 million. Stocks will be allotted to stockholders at one share per 80 holding stocks.

► **Ansett-ANA of Australia** has placed orders for additional tailspin equipment, including six Fokker F27s, two Vancos 555s and two new Lockheed Electras. Carriers previously has not approved additional Electra purchase but is expected to do so. Ansett plans tailspin service on all its routes to be able to land and a scheduled to take delivery of its first Electra in March with Vancos deliveries beginning in January.

► **Flight Engineers' International Assn.** has signed new contracts with the Flying Tiger Line and Seaboard & Western Air Lines. The two-year contracts involve about 147 flight engineers. Third-party-employment qualification must not arise in the negotiations because neither of the cargo airlines has ordered jets for tailspins.

► **Eastern Air Lines** has extended its contract with the CE BE Cleveland Co. Inc., Dayton, Ohio, for clearing repair, repair and assembly of integral fuel tanks on the airline's Constellation B-70. Contract runs on a job on 184-749 jets, 12 L-300s and options on 16 additional 945s at the company's Long Beach Airport facility.

# Airline Income & Expenses—May, 1958

(In Dollars)

	Passenger Revenue	U.S. Mail	Express	Freight	Charter	Total Operating Revenue	Total Operating Expenses	Net Income Before Taxes
<b>WESTERN AIRLINES</b>								
Revenue	\$8,455,936	945,006	9,867,149 <sup>1</sup>		18,279	26,330,499	24,444,196	2,237,474
Expenses	4,807,734	133,199	44,708	109,547	4,376	5,109,494	4,504,583	288,692
Regional	9,299,970	312,203	104,476	127,490	39,527	9,797,656	8,699,100	1,097,656
Continental	5,614,659	48,740	59,799	38,974	1,634	5,864,403	5,123,259	741,144
Dallas	9,426,107	161,291	75,268	234,174	40,419	9,940,959	8,807,818	1,133,141
Eastern	9,246,461	313,139	123,297 <sup>1</sup>	170,238	59,000	9,914,405	8,991,546	922,859
Midland	4,416,365	57,723	14,901	141,067	79,727	5,009,424	4,183,208	826,216
Northwest	8,614,322	108,740	69,447	178,197	43,476	9,044,424	12,126,420	-2,776,536
Western	4,463,282	121,206	215,549 <sup>1</sup>	171,190	11,190	5,008,199	5,408,140	-384,941
Trans World	10,517,913	209,810	848,172 <sup>1</sup>	30,437	37,977	11,704,149	10,708,649	995,500
United	11,595,992	831,158	1,747,087 <sup>1</sup>	10,430	18,480	26,464,158	25,641,638	822,520
Western*					5,523	31,974	794,414	-465,193
<b>WESTERN AIRLINES</b>								
Revenue	\$10,298	\$1,149	\$2,644 <sup>1</sup>			\$13,091	\$10,440	\$2,651
Expenses	499,711	7,120	37,222			544,053	544,053	-430,962
Continental	100,458	5,100	4,428			110,006	110,006	-99,590
Eastern	1,807,437	34,792	34,192 <sup>1</sup>			1,876,421	1,876,421	-1,781,974
Midland	226,109	8,416	1,617			236,142	236,142	-227,726
Northwest	1,701,844	331,664	336,276 <sup>1</sup>			2,369,784	2,369,784	-2,268,190
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# NASA Details Space Contracts Program

Washington—Proposals for a \$145 million contracting program will advance for space research and development are included in the supplementary budget request presented Congress by the late National Aeronautics and Space Administration.

It is complemented by a \$512.1 million NASA research and development program with other government agencies, such as USAR's Future Mission Division. Of the total \$157.1 million requested for research and development, \$117 million will be transferred to NASA from Defense Department accounts.

AVIATION WEEK is presenting below a detailed breakdown of the proposed industry contracting program at a glance to its readers. For a report on the overall NASA budget request, including the education and equipment program and salaries and expenses, see story on page 14.

## General

### Space Science ..... \$18,280,000

Design and development of vertical probe rocket vehicles .....	\$50,800
Research and development on scientific instrumentation for vertical probes such as temperature, density and spectrographic devices .....	700,800
Sensors for landing, tracking, data collection and analysis for vertical probes .....	1,750,800
Research and development on scientific instrumentation packages for satellites. Infrared radiance, cosmic rays, ultraviolet, X-ray radiation, magnetic fields and other geophysical phenomena will be investigated .....	3,000,800
Sensors for tracking and instrumenting space stations, and data collection and analysis for satellite experiments .....	3,210,800
Development of laser probe instruments, including sensor, scanner and specialized data playback and display instruments .....	700,800
Initial study of techniques for effecting landings of instrumented probes on the moon .....	200,800
Studies on long-range television systems particularly adapted to lunar probes .....	88,800
Initial studies of trajectories, orbits and vehicle design for interplanetary probes .....	1,600,800
Study of the performance requirements of vehicle-based astronomical instruments .....	100,800
Prototype satellite-borne astronomical site survey development .....	1,280,800
Command control and data handling and instrumentation for specific applications to satellite-borne astronomical instruments .....	1,200,800
Vehicle design and construction for astronomical satellites .....	800,800
Design studies on satellite booster vehicles .....	1,000,800
Design studies on upper-stage rocket vehicles for satellites .....	700,800
Analysis and studies on vehicle-based sensing devices to be used in probes and satellites .....	500,800
Studies of data handling devices of the type used in processing received experimental data .....	100,000

Studies relating to the development of tele-metricing services of the type used in the satellite program .....	200,000
Studies of precision requirements and system arrangements for range and range-rate systems .....	200,000
Design studies for interplanetary probes .....	400,000
Missile launch, repairs, alterations and space construction .....	100,000

### Space Technology ..... \$46,800,000

Initial development contract for a satellite-based rocket motor .....	\$18,800,000
Development contracts for high-energy rocket motors .....	4,900,000
12,000-lb. thrust fuselage-booster motor .....	2,120,800
80,000-lb. thrust fuselage-booster motor .....	1,000,800
100,000-lb. thrust fuselage-booster motor .....	1,000,800
Solid propellant rocket motor development contract .....	1,700,000
Studies of electric propulsion for space vehicles .....	200,000
Research and development on improved auxiliary power supplies .....	2,100,000
Design of specific power units .....	800,800
Investigations of engine motors .....	600,800
Development of generators, regulators and controls .....	800,000
Investigation of reaction controls for space vehicles .....	700,000
Investigation of singular momentum control for space vehicles .....	910,800
Refractive-index systems including investigations of rocket types, thrust processing and controls .....	500,000
Studies on gravity stabilization techniques for attitude control .....	700,800
Studies and prototype equipment development for the following phases of data handling and data transmission .....	1,500,800
Information coding and processing .....	800,000
Receiving components and systems .....	400,000
Transmitting components and systems .....	400,800
Extensible antennas for space vehicles and associated devices .....	600,000
General studies, such as the self-propulsion and of transponders, the improvement of analog-digital conversion and the development of data recording and retrieval techniques .....	100,000
Studies of the task requirements for space vehicles using manual controls .....	180,000
Studies of manual and computer display devices for use in space vehicles .....	450,000

Design studies for prototype control systems .....	250,800
Development of personnel equipment including in supply and material, waste disposal, emergency protection, and food supply .....	1,000,800
Research and development on guidance systems for space vehicles, including efforts as sensors such as gyroscopes and star trackers, accelerometer, reflexion analysis for specific sensors, and examination of experimental units .....	4,300,000
Landing guidance .....	1,500,000
Midcourse guidance .....	1,500,000
Terminal guidance .....	1,500,000
Landing control systems .....	400,000
Research and development on computers for vehicle-based systems and for the improvement of ground-based computers used to derive detailed orbital or flight path constants of space vehicles .....	1,700,000
Solid state components .....	800,000
Mechanical design and packaging .....	900,000
Logic design .....	900,000
Input-output devices .....	1,300,000
Reliability analysis .....	200,000
Instrumentation research and development to improve the sensitivity, range and reliability of sensing devices and techniques and to bring to an acceptable level of development these techniques and instruments which are required for scientific studies in space, but which have not yet been reduced to size and/or weight to acceptable values .....	5,910,800
Sensing devices for field, particle and radiation measurements .....	1,200,000
Sensing devices for biological and medical measurements .....	100,800
Sensing devices for determining the performance of the vehicle and its booster rocket .....	550,000
Data switching and internal transfer devices, such as magnetic or electronic tapes .....	550,000
Studies on methods and equipment for the measurement of vehicle position and velocity .....	850,000
Development of stabilization systems for a 100,000 lb. payload .....	1,000,800
Development of stabilization motor for a large capacity (more than 2,000 lb.) payload .....	500,000
Development of selection units for stabilized platform applications .....	1,100,800
Design work and adaptation of an IRHM booster for flight testing of high-capacity motors .....	1,700,800
Design studies and development of a multi-stage test satellite .....	1,300,000
Development of a solid fuel booster motor .....	500,000
Investigations of energy sources for use as auxiliary power supplies in space vehicles .....	200,000
Physiological and human engineering studies of the tasks involved in space vehicle operations .....	150,000
Studies in logical design for computers .....	100,000

Reliability analysis for computer systems .....	100,800
Design studies on switching and data storage .....	200,800
System analysis on position and velocity measurement techniques .....	50,800

### Materials and Communications ..... \$4,320,000

Development of technological satellite payloads .....	\$2,200,000
Heat balance and infrared spectrum units .....	400,000
Television-type dual-sensor units .....	1,600,000
Studies on advanced optical sensors .....	200,800
Communications service developments .....	2,820,000
Analysis of reflector sphere configurations .....	100,000
Studies of channel requirements .....	120,000
Studies of passive reflectors .....	200,000
Studies of active relays .....	150,000
Studies of electro-optical propagation .....	50,000
Design of a communications relay satellite .....	1,400,000

### Technology of Manned Space Flight Vehicles ..... \$6,500,000

Design and development of capsule units for manned space flight .....	\$1,700,000
Design and development of electronics units and instrumentation for manned flight capsules .....	1,000,000
Development and adaptation of booster units for suborbital and ballistic flight capsules .....	3,800,000
Total .....	\$75,800,800

## R&D—Supplies and Materials

### Space Science ..... \$27,420,000

Rockets used as vehicle propulsion units for vertical probes (approximately 30 vehicles—Juno and Astrobee) .....	\$3,000,000
Components and materials used in the assembly and test of the scientific payloads of the vertical probes .....	750,800
Booster conductors for geophysical satellite projects. These are modified IRHM units (m or eight units of Jupiter engine Type 15) .....	5,500,800
Upper stage solid rocket motors for geophysical satellite projects (eight modified Vanguard or Redstone vehicles) .....	6,500,000
Vehicle-based sensing devices for use in the geophysical satellite projects. This will include units used in measuring the performance of the propulsion stages and items within the payload .....	2,000,000
Material used in the assembly of test fixtures, calibration jigs and the like .....	300,800
Supplies used in assembling test bench versions of data handling devices .....	800,000
Components and materials used in the assembly of experimental and prototype measuring transmitters (includes units to be tested for measurement performance which will be expended in testing) .....	700,000



■ S. A. Rice (center) tests 25,000 hour Klystron with advanced Polaris test equipment at Defense Test Facility.

## Eimac Klystrons Going Strong after 25,000 Hours in Pole Vault Troop-Scatter Service

After 25,137 hours on the test, and still in perfect operating condition, this Eimac 2030,000F 11HF Klystron has been acquired through the cooperation of the U.S. Air Force and Canadian Forces, Ltd. This Klystron was one of the original tubes installed in Project Pole Vault, the first troop-scatter communications line ever established. The tube is just one of a number of Eimac Klystrons that have exceeded 25,000 hours of reliable on-the-air time in this system. Eimac Klystrons are used as both amplifiers in the Pole Vault 10 follow-on transmitting that handle multiple-channel voice and teletype communications. Experience with this first system in war early warning defense network confirmed Eimac's power-powered tropospheric scatter as an outstandingly dependable system of long distance communication.

The exceptional performance of these tubes under the difficult logistical and environmental conditions of the far north is indicative of the reliability and conservative rating of performance-proven Eimac extended-life Klystrons. Eimac amplifier Klystrons are now being used extensively for troop-scatter communications throughout the United States, Canada and other parts of the world. Eimac Klystrons for communications and pulse applications are now available covering frequencies from 10MF to 25HF and to multi-resonant output powers.

For further information, write for a copy of the 24 page booklet "Eimac Facts Don't Lie!"

**EITEL-McCULLOUGH, INC.**  
SAN BERNARDINO - CALIFORNIA

*Eimac First for reliable troop-scatter klystrons*



### Products Designed and Manufactured by Eimac

Negative Grid Tubes	Resonant Tube Assemblies
Puller and Amplifier Klystrons	Resonant Switches
Centric Receiving Tubes	Resonant Pumps

Includes the most extensive line of ceramic electronic tubes

Components and materials for the development of klystron tubes and vacuum tubes	400,000
Components and materials for the development of range and range-rate measuring systems	575,000
Booster rocket systems for lunar probes (two units of Thor and/or Jupiter type)	2,500,000
Upper stage rockets and assemblies for lunar probes	1,400,000
Materials and components for the development work on the lunar probe program	200,000
Materials used in initial investigations of the lunar landing problem	100,000
Optical, mechanical and electrical components used in developing a prototype and life-size astronomical telescope	200,000
Materials and components for the development of a command and control, data handling and transmission system for the astronomical telescope	200,000
Materials used in the development of a prototype satellite vehicle carrying an astronomical telescope	100,000
<b>Space Technology</b>	<b>\$17,300,000</b>
Experiments to be tested in the development of solid propellant rocket motors (new development)	7,000,000
Components and materials to be used in the development of electric propulsion systems for space vehicles	100,000
Components and materials used in the development of auxiliary power supplies to be used in the following areas:	
Specific payload design	100,000
Energy source development	100,000
Development of generators, regulators and power supply controls	500,000
Components and materials used in space vehicle and rocket control system development as follows:	
Reaction control systems	200,000
Angular momentum control	400,000
Retro-rocket systems	700,000
Components and materials used in development of data handling and data transmission systems for space vehicles as follows:	
Coding and processing studies	200,000
Receiving equipment research	200,000
Transmitting components research	200,000
Erectible microwave antennas	100,000
Other areas such as transportation, maintenance and recovery	200,000
Components and materials used in research and development on displays and control consoles for space vehicles	
For graphics used in task studies	50,000
For mapping system studies	200,000
For prototype and/or mockup consoles	400,000
Components and materials used in development of personnel equipment	100,000
Components and materials used in studies of guidance systems, as follows:	
For launching guidance	400,000
For midcourse guidance	300,000
For terminal guidance	300,000
For landing control systems	100,000
Components and materials used in complete research and development, as follows:	
Research on solid state components	200,000
Studies of packaging techniques	200,000
Research on logical design	100,000
Development of input/output devices	200,000
Components and materials used in instrumentation research and development as follows:	
For development of field, portable, and radiation measurement devices	200,000
For biological and medical instruments	100,000
For vehicle performance measurements	100,000
For data switching and storage devices	500,000
For transmitting components for range and velocity systems	100,000
For receiving components for range and velocity systems	100,000
For computer development	100,000
Components and materials used in research on stabilized platform systems	100,000
For light/laser platforms	100,000
For heavy-load platforms	500,000
For stabilization systems development	500,000
Booster rocket systems for high-energy launch systems test vehicles two units of Sergeant and Antares	2,300,000
Booster rocket systems and payload recovery vehicles for orbital test vehicles (two units of Thor and/or Jupiter rocket systems and retro-rockets)	3,300,000
Test vehicles for solid-fuel rocket development (two Sergeant or equivalent)	2,300,000
<b>Meteorology and Communications</b>	<b>\$5,700,000</b>
Components and materials for development of refined and test balance instruments	200,000
Components and materials for use in development work on the television-type cloud radar satellite equipment	100,000
Booster (and Thor Able) vehicle, and payload recovery systems for a meteorological satellite	3,200,000
Materials for studies of passive reflection for non-satellite altimeters	50,000
Components and materials for research on active radar	50,000
Booster (and Thor or Jupiter) and vehicle for orbital communications services satellite	2,000,000
<b>Technology of Manned Space Flight Vehicles</b>	<b>\$21,500,000</b>
Prototype test space capsule systems (eight test capsules to be expended in test)	4,500,000
Prototype capsule electronics and instruments (three test capsules for eight capsules to be expended in test)	1,000,000
Booster systems for sub-orbital flight tests (eight Thor Able or equivalent vehicles)	5,100,000

# for missile and rocket telemetry recording ...the Honeywell 3170 precision tape transport

a rack-mounted unit designed by the largest manufacturer of instruments and controls — to meet the most demanding requirements of analog magnetic tape instrumentation.

Examine these brief specifications on the Honeywell 3170 high-performance tape transport—designed to provide unparalleled flexibility and performance in accordance with IRIG specifications. Complete mechanical and electronic data are available on request—including application engineering assistance.

## Performance

Extremely low "wow and flutter,"  
Tape speeds available from 60 to 1½ ips ... selected by switch, without belt change  
Less than 6 seconds start time and 1 second stop time at 60 ips, with 14" reels.  
Fast forward and rewind.

## Flexibility

Up to 14" reels with NARTS hubs standard.  
Plug-in, interchangeable, record and playback heads.  
In-field changerover for tape widths of 1½" to 2".  
Search and control capstan, optional.

## Features

Precision-ground mounting plates assure perpendicular head mounting on changes of head stacks.  
Precision capstan drive assembly with multiple-speed hydraulic synchronous motor, operated from line, speed control servo, or precision-frequency power generator.  
Closed-loop isolation of tape path, with precision, non-mechanical tape tension control of both supply and take-up reels.  
Honeywell multi-track heads—record and playback—conform with IRIG specifications.



The complete Honeywell data acquisition system features direct, wide-band FM, multiplexed FM, or PDM data recording by means of plug-in amplifiers. Electronic components have low wow and wow drift with temperature and voltage changes. Signal-to-noise ratio exceeds IRIG specifications.

For detailed information, call your local Honeywell field engineer ... he's as close as your phone. **MINNEAPOLIS-HONEYWELL**, 19721 Hanna Street, Beltsville, Maryland.

# Honeywell



First in Controls

Booster systems for full-scale flight test (two Atlas or equivalent each)	5,360,000
<b>Total</b>	<b>\$71,910,000</b>

## R&D—Equipment

<b>Space Science</b>	<b>\$1,850,000</b>
Equipment items for research and development studies on vehicle-borne sensors	300,000
Calibration and performance test equipment for use in the development of vehicle-borne space sensor instruments	700,000
Data handling equipment such as magnetic tape drives and cable connections	150,000
Experimental telemetry receiver systems for laboratory use in developing instrumentation techniques	100,000
Experimental laboratory equipment for work on surge and surge rate measuring systems	100,000
Test equipment for experimental investigations of linear loading techniques	100,000
Equipment for experimental tests of command control and data handling and transmission as applied to autonomous vehicles	300,000
Equipment for use in mockups of autonomous vehicles	100,000

<b>Space Technology</b>	<b>\$4,590,000</b>
Research and development test equipment for work on auxiliary power supplies	400,000
Development of systems	180,000
Energy source studies	320,000
Studies of generation and transmission	200,000
Research and development test equipment for use in projects on vehicle control	150,000
Directive controls	180,000
Angular momentum controls	350,000
Retrieval control systems	100,000
Research and development test equipment for use in projects relating to data handling and transmission	600,000
Receiving system studies	200,000
Transmitting system studies	200,000
Releasable system studies	200,000
Research and development test equipment	

for use in studies of navigation and navigation systems	100,000
Equipment for use in setting up prototype control systems	100,000
Equipment for research and development tests of personal equipment	100,000
Research and development test equipment for use in projects developing guidance systems	600,000
Equipment for use in the development of linear loading units	100,000
Research and development test equipment for computer studies	600,000
Solid state components	100,000
Computer packaging	200,000
Logic design	100,000
Input output devices	300,000
Research test equipment for use in the development of instrumentation	1,100,000
Field, ground and radiation measurements	200,000
Biological and medical measurements	100,000
Vehicle performance measurements	200,000
Data handling and storage devices	300,000
Position and velocity measurement systems	300,000
Equipment for testing static platform development articles	500,000
Light payloads	200,000
Heavy payloads	200,000
Submarine tests	100,000
Equipment items for use in setting up systems test vehicles and systems test articles	400,000

## Meteorology and Communications \$480,000

Test equipment for selection type meteorological satellite payload	\$200,000
Test equipment for studies of passive reflection	30,000
Test equipment for studies of active relay satellite components	30,000
Test equipment for use with the advanced communications satellite	180,000
<b>Total</b>	<b>\$7,180,000</b>

## Company Keys Space Data to Civil Uses

Glendale, Calif.—Space Electronics Corp., Glendale, has been awarded the development of ground and airborne electronic equipment for missiles and space vehicles. One of the major objectives of the company, a subsidiary of Pacific Automation Products, Inc., will be to apply technology gained in the U.S. missile and space programs to commercial and civilian fields.

These projects fields include use of radar techniques in air-to-air and air-to-ground; present and other developments; development of electronic satellites as well as solid state navigation aids;

and meteorological surveillance systems for observation of language weather conditions.

Principals of the new corporation include Dr. J. C. Fletcher, president, and J. W. Levens, vice president. Fletcher previously was director of Electronics Laboratory, Ramo-Woolbridge Corp., with responsibility for development of guidance and control systems for missiles in USAF ballistic missile programs.

Levens was associated with Ramo-Woolbridge as associate director of Electronics Laboratory, responsible for the development of concepts, systems engineering and technical direction of KCBM and IRBM guidance systems, as well as project engineer for the Minuteman solid propellant IRBM.

## Ramo-Woolbridge Plans Research Unit

Ramo-Woolbridge Corp. plans to buy a 90-acre site near the Glendale Research Center in California's San Fernando Valley for construction of a multi-million dollar research and development facility. First use of the new center will provide space for some 2,000 scientists engaged in advanced administrative personnel in approximately 150,000 sq ft.

Advanced electronic research and development projects will be transferred to the new facility upon completion and space vacated at Ramo-Woolbridge's present location will be used for space-related laboratory.



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... confidence that you'll be able to take off on schedule

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hour corrosion test

Non-corrosive after  
900 hour test

The one time when you can least afford a battery failure is during takeoff. But when you have Exide aircraft batteries in commercial planes, you have dependable protection against the major cause of sudden battery failure.

In place of ordinary positive plate grid alloys, Exide uses Silvum,\* a new, patented alloy that resists corrosion up to 100% longer. While in other batteries corrosion often causes sudden failure, Silvum grids normally outlast the battery. In addition, experience proves Silvum stretches battery life, giving you more value for your dollar.

Eliminate this needless threat of takeoff delay from your operations. Invest Exide Batteries. For information, call your nearby Exide sales office. Or write Exide Industrial Division, The Electric Storage Battery Company, Philadelphia 2, Pa.

\*U.S. Patent

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## Research and Development Decline

Department of Defense has adjusted funds obligated for research and development to the 1947-48 level of purchasing power of the dollar in the three estimates to the right. They show that, as actual purchasing power, the research and development effort of the three military services has declined from a peak of \$3,179 million in Fiscal 1912 during the Korea war to \$113 million in Fiscal 1955.

	Obligations (Actual) (\$10,000 Counted)			Obligations (Adjusted to 1947-48 Price Index) (\$10,000 Counted)		
	For Military Programs	For R & D Programs	Total R & D	For Military Programs	For R & D Programs	Total R & D
<b>1945 Fiscal Year</b>						
Army	23	181	204	23	181.9	204.9
Navy	49	131	180	49	134.6	183.6
Air Force	35	171	206	35.2	143.5	178.7
<b>Total</b>	<b>107</b>	<b>483</b>	<b>590</b>	<b>107.2</b>	<b>459.0</b>	<b>566.2</b>
<b>1947 Fiscal Year</b>						
Army	40.7	338.3	379.0	44.6	312.9	357.5
Navy	45	204	249	50.2	234.2	284.4
Air Force	22.1	377.9	400	26.3	347	373.3
<b>Total</b>	<b>107.8</b>	<b>899.2</b>	<b>1,007</b>	<b>121.1</b>	<b>794.1</b>	<b>915.2</b>
<b>1950 Fiscal Year</b>						
Army	26.8	343.3	370.1	34.3	300	334.3
Navy	24.9	489.5	514.4	31.6	386.6	418.2
Air Force	100.9	338.5	439.4	83.6	315.7	400.3
<b>Total</b>	<b>252.6</b>	<b>1,171.3</b>	<b>1,424</b>	<b>199.5</b>	<b>1,002.3</b>	<b>1,201.8</b>
<b>1953 Fiscal Year</b>						
Army	24.6	395.4	420	34.9	345.6	380.5
Navy	24.1	411.8	435.9	31	359.9	390.9
Air Force	101.9	375.1	477	82.2	319.7	401.9
<b>Total</b>	<b>250.6</b>	<b>1,182.3</b>	<b>1,433</b>	<b>148.1</b>	<b>1,025.2</b>	<b>1,173.3</b>
<b>1954 Fiscal Year</b>						
Army	42.8	391.3	434.1	49	316.4	365.4
Navy	55.1	340.9	396	62.2	277	339.2
Air Force	67.9	399.9	467.8	63.6	336	400.6
<b>Total</b>	<b>165.8</b>	<b>1,132.1</b>	<b>1,297.9</b>	<b>175.8</b>	<b>930</b>	<b>1,105.8</b>
<b>1955 Fiscal Year</b>						
Army	26.9	340.1	367	34.9	279.7	314.6
Navy	27.9	399.4	427.3	42	273.8	315.8
Air Force	64.7	395.9	460.6	66	260.3	326.3
<b>Total</b>	<b>119.5</b>	<b>1,135.4</b>	<b>1,254.9</b>	<b>142.9</b>	<b>813.8</b>	<b>956.7</b>
<b>1956 Fiscal Year</b>						
Army	112.2	361.2	473.4	73.4	279.6	353
Navy	36.8	264.2	301	46.9	247.9	294.8
Air Force	45	467	512	28.3	307.9	336.2
<b>Total</b>	<b>194</b>	<b>1,092.6</b>	<b>1,286.6</b>	<b>148.6</b>	<b>835.4</b>	<b>984</b>
<b>1957 Fiscal Year</b>						
Army	129	383	512	88.8	307.1	395.9
Navy	146.9	379.8	526.7	91.9	297	388.9
Air Force	91.3	493.9	585.2	27.1	376.9	404
<b>Total</b>	<b>367.2</b>	<b>1,256.7</b>	<b>1,623.9</b>	<b>207.8</b>	<b>981.1</b>	<b>1,188.9</b>
<b>1958 Fiscal Year</b>						
Army	139.4	344.1	483.5	79.7	265.1	344.8
Navy	132.3	275.9	408.2	123.9	252.9	376.8
Air Force	104.1	454.4	558.5	85	388.9	473.9
<b>Total</b>	<b>375.8</b>	<b>1,074.4</b>	<b>1,450.2</b>	<b>289.5</b>	<b>906.1</b>	<b>1,195.6</b>

# Roylme

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## USAF Contracts

Following is a list of unclassified contracts for \$25,000 and over as released by U. S. Air Force contracting offices.

**RESEARCH INTERESTS AND POLICE SERVICE OF SCIENTIFIC RESEARCH, INC.** for research on development chemical weapons for U. S.

**Research at the University of California, Berkeley, Calif.** research on "Chemical Physics of Elementary Processes" LAF 49-0041-011, \$25,000.

**Research at the University of Michigan, Ann Arbor, Mich.** research and research on "Physics of Elementary Processes in Gases" LAF 49-0041-011, \$25,000.

**Research at the University of Wisconsin, Madison, Wis.** research on "Studies on the Complex Processes of Elementary Processes in Gases" LAF 49-0041-011, \$25,000.

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Republic Aviation Corp. Engineering Research Center will custom size separate laboratories for space research projects.

## Republic Launches Space Program

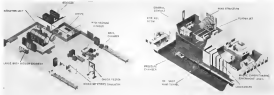
Farmersville, N. Y.—Republic Aviation Corp. is now working with contractors on the plans for its recently acquired 514 million Engineering Research Center (AW Aug. 4, p. 37) and expects to begin construction sometime this fall, probably in October.

Part of the company's four-year \$15 million program of research and development, which is aimed at getting Republic established in the space and space field, the new facility will house new laboratories for research and development and there for the study of materials and processes.

Space Environmental Development



REPUBLIC President Moody Pauls (left) and Vice President Research and Development Alexander Kordish (right).



SPACE facility (left) will have a sloped-chamber microclimate chamber. Re-entry unit (right) will have Mach 20 wind tunnel.







**INTELLIGENT** panel installed in autonomous warfare helicopter looks conventional, yet is coupled with a unique automatic flight control management and autonomous information processing derived under ANIP philosophy specifically for autonomous warfare missions.

## Navy Adapts Cockpit Display Knowledge

By Richard Sweency

**Johnsville, Pa.**—While long range objectives of highly integrated pilot displays are pursued in the Army-Navy Instrument Program (ANIP), Navy seeks to use knowledge gained in the program to put intrinsic advanced displays into operational production aircraft as soon as possible.

Aeronautical Instruments Laboratory at Naval Air Development Center has the task of rendering these displays, and can follow three main paths to accomplish its mission:

- Use knowledge from ANIP long range research which can be translated into production hardware within the time span allowed to develop a system for a particular forthcoming aircraft.
- Institute research efforts on its own to specific areas where it may well lead to hardware for which laboratory need is acute.
- Direct development efforts of the many contractors involved toward one particular system which is aimed at a particular aircraft.

Overall ANIP efforts are in three main research, development and application. Defining the terms and clarifying them as they apply to this program could yield that description. "Major research is seeking facts, trends and principles about man's relationship to an aircraft environment, which can be mechanized and evaluated as a basis of progressive steps, in systems, for employment in specific aircraft."

In the effect, general research work is directed by Douglas Aircraft El Segundo

Division in fixed wing aircraft, by Bell Helicopter Corp. in rotary wing aircraft, and is monitored by Office of Naval Research.

Development work is monitored and funded by Bureau of Aeronautics through its field activity—Aeronautical Instruments Laboratory at NAEC.

Application effort, also monitored and funded by Bellco, works through specific authentic contractors.

ANIP concept calls for showing pilot an analogy to visual (contact) flight to answer the phrase question "What am I doing?" "Where am I doing?" "What should I be doing?"



**HORIZONTAL** situation display part of a DV-4 panel has explicit reference to center, distributed points above and to side of horizon, giving aircraft position relative to destination.

Vertical, flat transparent cathode ray tube in front of pilot presents analogy to what he would see looking out wind screen while flying straight. Another cathode ray tube presentation on horizontal plane in front of pilot makes use of small airplane symbol and map to give analogy to what would be seen looking out bottom of plane of ground. These elements act to answer what and how pilot is doing.

Answer to what he should be doing is to be provided by a "pathway in the sky" to be presented on vertical tube. Pilot flies to the path.

Data for the overall presentation is to be gathered by various sensors aboard the aircraft, and information from these sensors is passed along to a central data computer in the aircraft. Computer results go to an electronic signal generator which changes computer output into proper signals to generate the desired indications on the proper indicator. Keyhole problems in the program has been the signal generator.

### Electronic Signal Generator

In April, Office of Naval Research announced purchase of an electronic signal generator developed by Kaiser Aircraft & Electronics Division of Kaiser Corp., which will electronically present the pathway in the sky concept (AW April 23, p. 53). Development of this generator has been termed a significant step toward ANIP final goal.

Kaiser developed the unit with one, four, and six tube units, as a concept, as a prototype. While the unit is very new, apparent capabilities indicate it may have an effect on several pilot



**REAR** rotary wing development phase panel, DV-2, shows pilot display basically similar to conventional panels, but warning and advisory tone generated is different. The DV-2 rotary wing effort has pulled operational equipment in the steadily HSS IN (AW June 26, p. 34).

## To Aircraft

display projects now in stages varying from proposal to flight test development hardware fabrication.

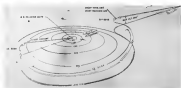
Since ANIP work, maximum staff in many cases, a method of identifying work progress has been devised, which uses three symbols, two letters and a dash number. First letter is either "R" or "D" denoting research or development. Second letter is either "V" or "I" for fixed wing or rotary wing respectively. Dash number identifies the phase, with Army using odd numbers. Navy uses "100s, RV-1 would identify research, fixed wing. Army, first phase DV-4 would specify development, rotary wing. Navy third phase.

### Current Efforts

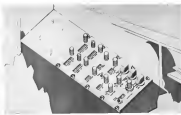
Current efforts in development area, including fabrication of components concerning three different phases which often are new or soon will be undergoing flight tests, were outlined in a paper by Louis Gaurico, systems engineer, Design and Stabilization Division, Aeronautical Instruments Laboratory.

Paper originally was delivered during closed session of ANIP symposium in Los Angeles last October (AW Oct. 21, p. 34), and recently has been declassified. Equipments described by Gaurico are Aeronautical Instruments Laboratory development units, installed directly in flat plate equipment at flight test at Douglas El Segundo, which is actually a research tool.

Two fixed and one rotary wing of three-DV-2, DV-3, and DV-4 units described in the paper. Another project based on lightly used DV-4B. The



**PROGRAMMED** autonomous warfare maneuvers could be performed by the steadily HSS IN helicopter with instrumentation which is a product of the DV-2 phase of the Army-Navy Instrument Program. Maneuver calls for knowing maneuvers over water while headed into wind, finding area, according to next point to repeat operation.



**DRAWING** shows controls for the DV-4 installation. In this installation, automatic programming in all flight modes is replaced by pilot manual input. The greatest apparent difference between the DV-2 and DV-4 units is in attitude display.

## Kaiser Builds Signal Generator

Oakland, Calif.—Recently announced electronic signal generator developed by Kaiser Aircraft Co. Electronic Division of Kaiser Corp. today came to one post-test model. However, company says, design is such that unit's capabilities can be expanded to meet any presently conceived requirement, plus proper loading and output.

Company has this in its report the generator which, although it was developed as a supporting unit, supplies the previously existing link in Army-Navy Instrument Program system chain of sensor-computer-designed generator-pilot display.

It will provide simplified, absolute signal conversion to pilots under any weather conditions. It is flexible, adaptable to all types of aircraft. It can provide with information and their time (1 sec. to 1 min.) information which pilot needs. It provides 1-1000, backward (as in helicopter flight), forward and vertical motion. It can accept inputs in form of voltage or shaft motion, from any computer which presently is available in ANIP effort, and will be able to process any data these computers might handle.

It can accept inputs from today's navigation aids such as VOR, ILS, etc. low frequency beacons, with proper adaptation. It can accept inputs from today's vertical and directional gyro as well as from altimeter gyro system, inertial stable platform. It has flexibility in the type of information it can present to represent work, day, other data which would be shown in this display.

Kaiser says that within one year's time, again given proper loading and support, generator's value would be:

- Completely unattended unit.
- Package of less than 1,000 cu. in., a 500-watt motor, less control size.
- Weight of less than 2 to 3 lb., a 500-watt motor, less control size.
- Capable of providing pilot with wide angle of view and wide.
- Flying in high performance aircraft.

Although degree of complexity of the information to be processed would effect development time cycle, company says it feels generator can keep pace with all other elements used in ANIP display.

tion, V is diagram (plot of velocity and acceleration) with curve indicating aircraft's structural limitations at all altitudes and speeds because integral part of flight director and command system. These flight director requirements thus formed the design objectives of the control computer.

Computer involved is termed "Automatic Integrated Director Equipment," or AIDE computer, and its main operations function is computing flight director information.

Ideally, Casasco says, it should be able to compute flight director information from continuous and automatic programming, but such a computer does not as yet exist for airborne applications. For DV-1 work, programmer was selected whereby pilot manually inserts protection of existing flight conditions in place of automatic programming, after which computer serves it desired flight path.

Programming is accomplished by guiding, tabulating and command translation techniques developed for this phase from earlier studies conducted by Naval Research Laboratory. These programmed signals are then converted by a mechanism of the V-a carriage so that optimum speed is continuously maintained, yet structural limits of aircraft are never permitted to be exceeded. Programming is accomplished from a console.

Computer used for this phase is, Casasco says, effectively a three-dimensional desk rotating navigational type computer providing the position and speed coordinates when plane is not a three status (GCI or vectoring) information is used in less than aircraft

DV-1 and DV-4 equipments were merely strapping stores in hardware, with specifications decided as being within bounds of existing accomplishments within time.

The DV-2 relay wing effort has yielded operational equipment in the H88-1 hydrogen, an anti-reference within machine.

DV-2 involved modifying current instruments to yield a display on ANIP principles. Initiated in November, 1954, project had a time interval of approximately one year, due to early requirements of a high visibility display with good controlled output capabilities. Equipment was to be contained in a TTV-1 aircraft.

First step was tabulating existing requirements, and situation and command information pilot needs were tabulated for normal and intercept modes. In this, status and reference information were considered. Basic TTV-1 equipment was analyzed for human engineering aspects.

### Task Control

Flight director and command element of system was based on fact that any aircraft flight path is determined by its vertical and horizontal tracks with respect to a reference, such as earth. Control of these tracks to achieve a desired path is a function of speed, direction, course, speed and vertical

speed are only constraints required for all flight modes.

Real problem of flight command, the paper states, lies in establishing the desired path and determining optimum method of achieving it within structural and power limits of aircraft. Airframe dynamics make all flight con-



DV-1 PANEL, which built the today's punch yet gets its information from Army-Navy Instrument Program philosophy source and now flows through a computer, is devoid of duplication of paper or hardware. Lower right page on displays for emergency operations.

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a used in intercept mode. These dual-sensor instruments are continuously computed and stored data aircraft in a GCI or vector mode, for use as required.

Several screens are introduced in DV-2. Control via reference vectors, control via data system are basically new, and present data functions is applied to aircraft as a status information generator and for the intercept, the AN/USC-3 Dynamic Address System is made a part of the vehicle.

While the instrument panel resulting from this technology does not show a virtual duplicate from today's display, it is not a duplicate, provides the "base aircraft" objective of the research effort using generally available equipment.

## Spatial Orientation

Spatial orientation is displayed in conventional manner, and cross position command the direction to be followed. Position with respect to destination or target is shown on central display on distance and bearing. Time to go to destination or target is shown as a number. Below display is a row of instruments which are self-contained standby gauges, and warning, control or emergency conditions.

DV-2 panel now is being installed in TIV-1, with flight equipped to start in August.

Research the DV-2 step was taken, since it provides little essential difference in what pilot sees, at first experiment behind the panel needs development as much as what the pilot sees. Control panel, and warning, control or emergency data, needs to be solved in being hardware as well as in theory and on paper.

In addition, system engineering required, was one to five applications needed to be used, the needs tested, before the basic philosophy became a function for better work. Otherwise, an undetected flaw might show up in several steps in the future, detecting error process went out of focus and error was observed and proper remedy applied.

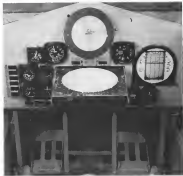
With DV-2 will under way, DV-4 was initiated in March, 1957.

## Alt-Weather Interceptor

Requirements were for alt-weather interceptors, including technique, GCI, Low Altitude Bombing System (LABS), and altitude intercept. Fluid management was to be included in this effort. Appropriately a case your time limit was applied to this effort.

DV-3 bases have been relatively well placed out, they were used as starting point for DV-4 work.

It was indicated that the following could be incorporated into the display:  
• Altitude intercept table could be



DV-4 control panel, as it would appear in an operational airplane is shown in a mockup.

integrated into the system at the lock-on point.

• Flight range and endurance data could be generated all along the mission profile to that point of return could be determined.

• A computer known as LABSAC which constantly receives necessary for an identification point could be included in the system.

• Expanded navigation capability based on great circle navigation, and not based by any latitude restrictions, was added.

Computer, based on three requirements, was to stay at some 25 lb weight and 1 cu ft volume as had been planned for the DV-2. It was to be operated from a console again, with pilot manual inputs in place for the unavoidable complete automatic progress.

Logic separate difference between DV-2 and DV-4 units in its altitude display, with control analog development authorization for DV-4 on two systems.

One approach was electronic display generator for outside air table presentation, the other was electro-mechanical. At first DV-4 was underwritten, it was determined that an electro-mechanical unit could be fabricated within the time limit, would be electrically and mechanically interchangeable with the control display in the DV-3 panel.

Series equalizer was unique at this time, and modifications were that

this type display would involve weight and volume penalties which were not worth incurring, so electro-mechanical system was chosen.

Electro-mechanical display has grill face, with signs moving toward pilot in a function of ground speed, signs moving laterally in direction of turn rate, and closed mass pattern above horizon to add more motion effect when turning.

Flight director display in form of a plumb, supplemented over other attitude information, in closest approach to "path in the sky" concept.

## Relative Position

Horizontal reference display presents to pilot his position relative to a destination or target, with the objective shown as an illuminated circle, yielding qualitative position information. The circle bearing can be read against an active heading ring or perimeter of display, while distance in miles and time in minutes to destination can be read off six counters. A scale changer is provided for the display.

DV-4 installation is being accomplished in a Grumman F9F-51. A computer table installation is being accomplished, and plan is perhaps six months from flight test.

As in the DV-2, computer is based to determine of particular distance, DV-2 the TIV-1 and DV-4 for F9F-51. Both computers could easily be reprogrammed for any other distance. In

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One of the primary objectives of the Army's growing aviation program is to execute its many varied missions with the fewest possible aircraft types. The necessity of this consolidation in terms of logistics and over all economy is underlined by the continued increase in the number of missions to be loaded on the nuclear battlefield. The Army's H-23D "Rover" meets this multi-mission requirement in the light helicopter category. Its versatility is obtained by a performance, a load and space capacity, and a responsiveness to perform its tasks without compromise on land or in the air.

Hiller has now installed a new 325 hp engine in the basic H-23D ship. The resulting 128 aircraft—the most powerful in the two to four place class—is prepared to capture a whole new order of missions, many of them not previously considered within the realm of light helicopters.

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**ATTITUDE** display portion of DV-4 instrument panel now plans to include "pitch in the sky" concept, showing pitch limits in traditional ground speed, and variation of apex of path to indicate turning. Unit is electrically actuated.

both cases are of the new equipment which indicate gross weight of the aircraft according to G-meter, leading to better surface characteristics.

Following DV-4 next step will probably be designated DV-50, and mission will be a low-level attack-type. Here, the paper states, a great deal will depend on airborne equipment raised and its method of presentation. The value of this aircraft must be judged by adequacy of the pilot's environment.

To get best results, situation, command and aircraft status will be carried forward (although known generator aircraft outside will make electronic target system unable instead of electronic). New sensor equipment for this particular mission will be added. New sensors will include a terrain clearance radar and a track while area radar. Computer will further integrate the screen and compute flight path from a programmer which considers all the sensors.

Investigated and found feasible for DV-10 are:

- Addition of a screen motion actuated from flight path signals to provide cross plane attitude flight throughout all altitudes, speeds and attitudes, with full acceleration from straight and power viewpoints.
- Computation of flight path from terrain clearance radar in accordance with performance instructions (programmer's ring).
- Use of signals from track-while-area radar to program path according to either strongest target, nearest target or most feasible target.

DV-10 work will bring forth equipment which would be at the level required for the proposed Douglas A-10 production concept (A-10, Dec. 10, p. 56). Should there be a direct coupling

of the DV-10 equipment into the A-10 program, designation would be AV-10, indicating attack-type, fixed-wing, Navy, fifth phase. In the AV-10 phase, the program would be carried through flight test and evaluation of the equipment for fulfilling mission requirements in A-10. Following this, ANF development would disappear and project would become purely a production matter like the rest of the airplane, its sensors and components.

However, it is emphasized that there is no direct link between the DV-10 and A-10 proposed at this moment.

Considering the mission, it is more likely that a cockpit based on DV-10,

with electronic pack as the sky, will be proposed by A-10. For the forthcoming Congress low level attack plane, although there has been no official indication of this step.

Should DV-10 be installed toward a production airplane, the cockpit can be presentation as in the A-10 proposal must probably would be used, which is a conventional display table rather than flat and irregular.

Also, instruments are possibly that while the path in the sky could be used to present speed commands, the DV-10 equipment would probably use a conventional approach. Each step of progress with cockpit in completed state

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rather than new information. A new vertical-approach altimeter would most probably be used. Engine gauges would be conventional, and standby emergency gauges omitted.

Disoriental situation display would probably be the fixed airplane-viewing map type, with a range scale indicating steel and digital readouts also provided.

Illustrative of immediate possibilities, General's paper outlined work done with HHS-1 helicopter, anti-submarine warfare craft, in investigating engine, pilot, medium conversion compatible to a fixed wing ILS at GCA every 5 min for a 4-hr period.

Underlying all DRI-2, the development effort outlined requirements, integrated system and developed an electrically prepositioned instrument Visorair cuffs for hovering, center line with respect to water at present altitude of 30 to 50 ft while headed into wind, and dump the visor, then proceed to next point and repeat the operation.

Computer programmer developed for the mission, called Helicopter Integrated Director Equipment (HIDE), has provision for long range navigation to target area, localized navigation in target area. It utilizes dual receiving navigation techniques with Doppler correction for wind and prediction of coming, power, cruise altitude, cruise speed and vertical speed rates. It computes flight path and issues director command signals for cyclic and collective control of aircraft.

Unit serves both flight director and automatic controls.

In automatic controls, the descending mode is separated from attitude stabilization mode so that dropping or stable air separation is integral with level control surface, and attitude stabilization is superimposed in parallel to the control through a system control stick steering arrangement. With this relief can override the system without any appreciable change in feel. This enables to cruise as well as collective control. Engine speed also is governed automatically and can be overridden by manual, throttle action.

Although not the configuration of the DRI-2 pilot display model, the installation based on DRI-2 has caused much to be demonstrated 1981 N.

DRI-2 panel itself provides complete instrumentation for pilot and cockpit, and while conventional expert handling conventional, they say, like the DRI-2, a team from which future work can proceed.

An additional part of the ANP program, a technique for assessing and evaluating human performance in aircraft has been developed, and equipment will be installed and used in the DRI-2 flight test program.

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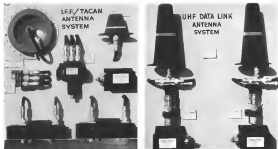
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INTEGRATED antenna system automatically select signals at belly antenna which provides optimum signal at the moment, the cable Tacan and UHF equipment to share a common pair of antennas and UHF and data link to share another pair.

## Airborne Avionics Share Antenna System

Los Angeles—Avionics systems which provide more reliable communication, navigation and identification signals with only half as many aircraft sockets as previously, are now being produced here by Electronic Specialty Co. for use on all Century Series aircraft. One antenna system is used with Tacan and UHF (identification forced or free) transponders; the other serves as an airborne communication receiver, transmitter and data link system.

### "Shedding"

In most twin aircraft have employed two antennas, one situated atop the fuselage or tail, the other in the belly, to get around the problem of "shedding." Shedding occurs when portions of the airplane's ground-based antenna and the ground station as frequently happens during aircraft maneuvers.

With two antennas, the danger of shedding is reduced but another problem arises which antenna should be connected to the receiver-transmitter at any instant and when should the choice of antennas be changed. Systems which Electronic Specialty produces are designed to automatically solve this problem.

By means of two diode tubes, it is possible to operate both HF and Tacan equipment from a single pair of antennas (topside and belly) since both equipments operate in the 1,000 mc band.

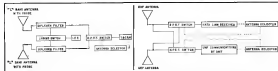
To assure that ground-based HF interrogation will obtain a prompt reply from the airplane's HF transponder regardless of aircraft attitude, the antenna employs a latching switch which alternately connects the transponder to the belly antenna, then to the topside antenna, then back to the belly

antenna at a rate of 30 cycles per second. The dwell time in either position is sufficient for the antenna to receive HF interrogation and to make reply.

### Latching Switch

Latching switch developed by Electronic Specialty employs special lubrication of contacts and bearings to provide long, reliable service. An F-100 acquires the switch to pass a 1,000-psi life test, at 75 million cycles without failure.

A signal-locking arrangement is employed for selection of the optimum antenna for Tacan operations. Cycling switch transfers the Tacan receiver back and forth between the same pair of antennas used for HF (isolated through diode tubes) at rate of 75 cycles per second. Selector picks the antenna with the strongest 30° signal and stops cycling.



BLOCK diagram of Tacan/UHF antenna system at left; UHF/data link system at right. Systems reduce number of antennas.



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Dr. Miller has been engaged in a wide variety of advanced engineering projects for 35 years. At General Mills he directs and coordinates the efforts of an excellent staff of unusually capable engineers. Dr. Miller is also more than just an engineer... he General Mills way of life.

When the RF signal drops below the preset value, the selector switches to the other antenna.

If no signal is available there, cycling continues until an adequate signal is obtained. During the cycling operation, the cockpit assembly indicator is automatically disabled to prevent its giving an erratic bearing indication.

### UHF Communications

Similar arrangement is used for UHF communications act and data link operation, but a different pair of antennas is employed because of the different frequency band. Automatic cycling operation stops when one of the two antennas provides sufficient RF signal to sustain the receiver speech circuit relay. When the signal drops below the preset threshold cycling automatically starts.

If the pilot wishes to make a voice transmission while the cycling operation is under way, selecting link of sufficient RF signal from either antenna, the Electronic Speech system "re-transmits" which antenna provided the first usable signal and automatically converts the transmitter to that antenna when the pilot pushes his microphone button.

Data link equipment has its own cycling switch and antenna selector, designed to lock the system signal antenna independently of the UHF communications antenna selector. Both can share the same antenna or operate



### F-27 Gear Control

Integrated landing gear control used in Fairchild F-27 includes individual gear up-down indicators, wheel speed control handle which is held in down position by released wheel weight to third down landing gear in present inadvertent situation on the ground. Warning light can be checked in pilot when desired. Extra guard is automatically locked in accordance with MIL-PT-7765. Device is made by Avcon Products Engineering Corp., Davis, N. J.

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from different cut-work, depending upon the particular situation. However, if each time as the pilot is using his GUDF transceiver, the data link set is automatically connected to the other antenna to provide at least minimum service, rather than disabling the data link transmission during.

The single-pole double-throw (SPDT) coaxial antenna switch, developed by Electronic Specialty, weighs only 5 oz., has a switching time of 18 milliseconds. The two antennas and each have 6.1 amp at 28 v.d.c.

The HF/Tenax diplexer slices each weigh 14 lb and measure 4 1/2 x 2 1/2 in., considerably smaller than former equivalents, according to the company. Electronic Specialty Co.'s address is 5121 San Fernando Road, Los Angeles.

## Summers Gyroscope Ends Merger Talks

Los Angeles—Plans for proposed merger of Summers Gyroscope Co., Inc., Santa Monica, Calif. and Norcon Corp., (Aircor Corp. subsidiary), have been terminated. Both companies felt that it was not feasible or practical to merge because of the size of the Summers deficit. Company officers felt that Norcon shareholders would not approve the required agreement.

Discussion with the parent Aircor Corp. culminated in a \$150,000 loan to Summers Co. in 6% interest, due June 15, 1983, with proposal for an additional \$425,000 loan in periodic payments to be completed by first week of October, 1983.

Under January agreement between Summers and Norcon, Summers was loaned \$154,300.65 on 6% semi-annual promissory notes, principal and interest due March 3, 1983. These notes will be convertible into common stock of the company at any time prior to three months on basis of \$1.58 per share, adding such notes at face amount plus an accrued unpaid interest.

Also the loan provided to use the efforts to obtain additional financing, not to exceed \$750,000, on terms under which entire amount of additional financing would be furnished and convertible into common stock of the company on same basis, including conversion rights and options. However, there is no obligation of Aircor to provide or Summers to accept such financing.

Summers recently moved into its new 300,000 sq. ft. plant which utilizes continuous dust free controlled atmosphere conditions for testing precision instruments for aircraft and missiles.

Air Force has awarded Summers a new \$1,700,000 contract for improved vertical gyro indicators system which is smaller and lighter than previous model and gives added performance.

## AMB Contracts

Washington—Always Modernization Board obligated over 47% of the \$58.1 million appropriated for its activities during the year of its activities, including 46 of 24 major development programs and studies. List of AMB programs now under way, assigned objectives or development agencies, and amounts of the contracts follow:

CONTRACT	PROGRAM	AMOUNT
Airborne Instruments Laboratory, Inc.	Analysis of air traffic in the New York coastal area	\$140,000
The Franklin Institute of the State of Pennsylvania	Operational analysis of air traffic demands and delays in the New York ARTCC area	30,000
Naval Science Research Inc.	Analysis and report on lowest requirements applicable to airport lighting and marking systems	40,000
United Research Inc.	Method for determining economic value of AMB program with application to landing system program	44,000
University of California	Investigating and establishing configurations of high-speed cell test ways from cameras	44,000
Airborne Instruments Laboratory, Inc.	"Globe viewer" engineering and software architectural services	55,000
Aircraft Instruments, Inc.	Monitor for use with GFL data processing and display system	1,404,000
General Aeronautical Laboratory, Inc.	"Globe viewer" engineering and software architectural services	35,000
The Franklin Institute of the State of Pennsylvania	"Globe viewer" engineering and software architectural services	17,000
International Business Machines Corp.	First-time simulation and statistical data	37,000
General Precision Laboratory, Inc.	ATC data processing system	6,345,000
Radio Corporation of America	Experimental system to ground air-ground communication system	1,404,000
Airborne Instruments Laboratory, Inc.	Experimental use of hypersonic navigation system for helicopter	180,000
Bell Helicopter Corp.	Modification of M-130 helicopter with special instrumentation and flight testing of AMB program	68,000
Smith Aviation Corp., Pacific Division	Study of Bendix-Pacific navigation system for experimental helicopter operations	175,000
ARL, Pullard and ARL	Survey, evaluate and report on site for AMB program research center	6,000
ARL, Pullard and ARL	Preparation of plans and specifications for construction of 12,000 sq. ft. research facility	\$123,000
New York Airways, Inc.	To fly and report on performance of Bendix-Pacific airborne navigational equipment	40,000
	Total—Private contractors	10,499,000
Department of Commerce, Civil Aeronautics Administration, Technical Development Center	Miscellaneous projects for the Director of Systems Analysis and Development	\$1,138,000
Department of the Navy, Chief of Naval Research	Research reports work in problems of ATC	40,000
Department of the Air Force	Feasibility design parameters for landing control aid	15,000
Department of the Army, House of Representatives	Miscellaneous projects for the Development Division	360,000
Department of the Navy, Bureau of Ships	Modify sensor and navigation and develop sensor video test equipment	85,000
Department of Commerce, Weather Bureau	Operate weather sensor at NAUTC	5,000
	Total—Other Government Agencies	1,602,000
	GRAND TOTAL	14,331,000

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### B-58 Avionics Gear Tested on KC-97

Avionics gear systems of Convair B-58 Hustler bomber is fitted to nose of Boeing KC-97 tanker, which will be used as testbed for Hustler's navigation equipment. Testing is being done by Sperry Gyroscope Co. at Air Force Flight Lab, FL.

## Expansions, Changes In Avionics Industry

The Remo-Woodbridge Corp. has reorganized its major divisions and assigned top-level responsibilities, following recent announcement of plans to merge with Thompson Products (AW Feb 7, P. 28). Nine shifts include:

- **Dr. Ralph R. Johnson**, formerly vice president for research and development, becomes group vice president in charge of the General Electronics Group of division, responsible for research, engineering, manufacturing, and administrative activities of the company except those of Space Technology Laboratories and the Thompson-Remo-Woodbridge Products Co., both of which report directly to the president.
- **Dr. Burton F. Miller**, formerly vice president in charge of Communications Division, becomes vice president in charge of advanced systems planning.
- **William E. Mink**, formerly vice president in charge of Control Systems Division, becomes vice president of engineering, responsible for Control Systems, Communications Division and the Electronics Information Office, as R/W division located in Denver.
- **Irvia A. Bland**, vice president, manufacturing, will be responsible for these activities at both Denver and Los Angeles facilities.
- Other recently announced changes and expansions in the avionics field include:

- **Adams-Rendall Co., Inc.**, Cambridge, Mass., a name of newly formed company that will engage in electronic research, development, consulting and manufacturing. Address: 282 Main St.
- **Pyrocontrol Co. of America**, Portland, Maine, a new firm organized to carry out electronic instrumentation, thermocouples and accessories. President is John V. Meryem. Address: 640 East Lincoln Highway.

- **Stanford Research Institute** has formed a Radar Acoustics group within its engineering division to conduct applied research in radar navigation. Dr. Myron H. Logan heads the group.

- **General Controls Co.**, Cleveland, Calif., will build \$2 million expansion of its manufacturing facility. First part of plant will be 70,000 sq ft. with plans for future expansion to 250,000 sq ft.

- **Ilmor Industries**, Glendora, Calif., has formed new Avionics Electronics Division which will produce miniature transmitter potentiometers. George Elliott heads new division.

- **International Electronic Research Corp.**, Burbank, Calif., has formed two effort companies to manufacture its heat dissipating tube shields. Gerd Merschanz Corp., Chicago, will make and sell tube shields through an U.S. Europe. Leo Gloria, President of Pico, Texas, will produce and sell shields in France, West Germany, Switzerland, Holland, Belgium, Luxembourg and French possessions.

## FILTER CENTER

- **Lowfield Wash-Make**, a mobile digital computer mounted in a standard 35 ft. truck, is being developed for the Army Signal Corps by Silvaco Electronics Products, Inc. to solve a wide variety of military problems ranging from battle strategy to logistics. Company will be entirely manufactured and designed to function under battlefield conditions anywhere in the world.

- **Naval Conference**—There will be no formal papers delivered at Electronic Industries Ann Conference on Reliable Electronic Components, to be held Dec. 2-4 at the Sheraton-Hilton hotel in Dallas. Technical papers will be available at least 30 days prior to conference, according to present plans. Authors will merely give a 5-minute summary, then devote remainder of time to questions from the floor.

- **Signal On Dotted Line**—Major contract awards recently announced by various manufacturers include:

- **Allen B. De Mott Laboratories, Inc.**, 5711, 485, for establishment of technical laboratory at Avionics Modernization Board's new National Avionics Facilities Experimental Center in Atlantic City.

- **Brooks Radio**, more than \$1 million from Civil Aeronautics Administration, for automatic radio navigation and audio system to be installed on recently acquired CAA aircraft.

- **Phelps Corp.**, 51 million contract from Army, for installation of two radio battery extensive communications systems between Nike missile sites and anti-aircraft operations centers in Alaska.

- **North American's Avionics Division**, 300 million Air Force contract, for space parts for MG-4 for control system and in F-408s, employed by four NATO countries.

## NEW AVIONIC PRODUCTS

### Components & Devices

- **Silicon voltage references** with temperature coefficients to 0.001%/deg. C



are available in an airtight package. Reference voltage is stable over the operating range from -55 to +100°C. Reference constant of bromine/silver sealed glass diodes capable of operating in air moisture. Transistor Electronic Corp., Wakefield, Mass.

- **Large carbon tube**, Type WL-7106, is designed for extreme environmental conditions of shock, vibration, temperature and humidity. Tube will withstand vibration with 100's acceleration up to 500 cps, as specified in MIL-E-8271A. AG's certification (50 to 500 cps) tube shows heretofore analyses of it.



least 550 lines with 0.01 foot-candle fluxes of the photoresistor. Characteristics of 160's does not require subsequent operation. Tube is sufficiently sensitive to provide 250 line horizontal resolution with 0.0003 foot-candle as minimum of the photoresistor. West Electronics, Elmhurst, Ill. Division, P.O. Box 256, Elmhurst, N.Y.

- **Cost modules** allow semiconductor and passive security packaging densities to 740's. Instrument modules are placeable and replaceable. Used with printed circuit boards, they provide wa-



ing in three dimensions. With low power outputs, modules can be integrated with thermal leg packaging, which when wrapped together they can withstand stress shock. Lind Corp., 1181 Hughes Drive, Trenton, N.J.

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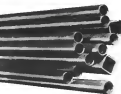


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## AERONAUTICAL ENGINEERING



General Electric J79 jet engine can develop 15,000 lb. thrust with afterburning, compared to J47s, 7,500 lb. in same weight class.

### J79 Uses Built-Up Parts to Save Weight

By J. S. Bantz, Jr.

Everdale, Ohio—General Electric J79 turbojet, which is powering most of the Mach 2 aircraft in the U. S., has set a new standard for lightness, primarily through the extensive possible use of built-up sheet metal parts to cover the main engine loads rather than the more traditional castings and forgings.

Record of these built-up parts is so strong as to date his old-fashioned General Electric's construction that the light type of construction would afford or little operation while providing new high thrust-to-weight ratios and correspondingly good aircraft performance. Basic engine weight of the Phase I J79 is about 3,100 lb. thrust with afterburning at around 15,000 lb. This thrust-to-weight ratio of about better than 10 clearly explains the low performance of such Mach 2 aircraft as

the Lockheed F-104, which presently holds both the world altitude and speed records of 91,291 ft. and 1,494.19 mph; the Convair B-58 which outruns most fighters; and the Chance Vought Republic F-104.

Newest aircraft such as the North American A-7 and the McDonnell F-4H are the improved Phase II engine, which is roughly the same weight as the Phase I but has about 65% better thrust and specific fuel consumption. This is primarily due to an increase in turbine diameter of about 2 in.

#### Total Running Time

Total running time on all data numbers of the J79 is now well over 37,000 hr., including more than 10,000 hr. of flight time. Indicators of the type of use this has been, and the wide application that the engine has found, is the General Electric report that an engine due in less than three years of ser-

vice to the air at the same time over the U. S., all running over Mach 2 and all J79 powered. This was the Grumman F-117, Convair B-58, McDonnell F-4H, Chance Vought Republic F-104 and the Lockheed F-104.

On the maintenance side, the high data numbers of the J79 which have over the last seven years an approval time between overhaul of 300 hr. The main reason required to accomplish this overhaul are difficult figures to arrive at because most overhauls include a good deal of rebuilding. Also, the experience of the facility performing the work is an important factor.

However, some of the civil firms that are bidding for the engines' jet or turbojet business are being their cost estimates on 500 man hours to overhaul the J79.

Basically, the design philosophy behind the J79 is similar to that being



Variable stator vanes are adjusted by General Electric technicians (left). Exhaust area petals open or close (right) for maximum thrust at all speeds and altitudes. Petals close the petals in closed position.



Exhaust area petals open or close (right) for maximum thrust at all speeds and altitudes. Petals close the petals in closed position.



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**RATIO** of turbojet diameter is much larger on J79, allowing greater inflow and exit flow. Length is 226.5 in. for J47 and 207 in. for J79 engine.

used today by General Electric and the Air Force in the development of the J47, Mach 3 engine. That philosophy is to make a new engine with one particular advantage.

The J47 is being designed specifically for use in the North American B-70, and the J79 was intended for the B-58. In its inception to give more engine performance to that engine in a combat situation. The design approach calls for close cooperation in all phases of the design between the engineer and powerplant manufacturer and certainly does not rely on test, as engine is good advantage in other aircraft.

### Many Philosophy

The Navy is using a different philosophy in designing a Mach 3 engine. It is using Pratt & Whitney carry the J47 will enter the testing phase before contracting for an advance to use the engine. This philosophy is to take the engine's capabilities firmly before relying upon it in designing a weapon system. It would prevent the over-engineering of the engine system when the Navy's first group of turbine aircraft were about all designed around the J47. The engine failed to meet its predicted performance.

These primary requirements for the J79 were stipulated in 1952 so that it would provide the proper power for the B-58 at all design speeds. These three requirements remained more or less fixed throughout the detailed negotiations between General and General Electric that resulted in a viable weapon system. They were for the J79 to provide:

- Minimum thrust-to-weight ratio of 5.0 to give the aircraft the desired speed, altitude and acceleration performance.
- High thrust at Mach 2 at 31,000 ft. and above.
- Efficient subsonic cruise at about Mach 0.9.

The engine also was to be ready for production in 1955.

To meet the Mach 2 specification a maximum air flow of about 165 lb per sec. was needed, and this meant using a very high tip-to-tip ratio compressor if the engine frontal area was to be adequate result. The engine also would require a large compressor of 17 or more stages to get the compression ratio of 12 to 15 needed for good fuel consumption at subsonic speeds.

In selecting the type of compressor, General Electric considered two conventional and separate studies of dual rotor and variable stator designs. The other known method of obtaining proper compressor operation at both supersonic and subsonic speeds, using pop-out valves on the rear compressor stage to prevent compressor stall and forward flow of air out of the compressor was judged to waste power and to be technically infeasible.

The variable stator approach was chosen because it indicated a considerable weight saving through use of one shaft and three main bearings in comparison with two shafts and nine main bearings with the dual rotor. Final compressor design has a 17-stage rotor with 19 stator stages. The first six stators and the inlet guide vane are variable.

Need for only three main bearings



**STAMPED** J79 rotor (left) and compressor with solid forged blade at inlet.



## Russians Show Vertical Takeoff Rig

Flight demonstration of Russian "Vostok" vertical takeoff rig was made at Moscow's Tushino airport (APR 28, p. 24). Vehicle was designed by A. Kopylovsky, a descendant by single inheritance which blends flow of air to control gas on four legs.

Under it possible to not only three fuselages in the engine, a front frame ahead of the compressor, a compressor rear frame and a turbine frame with each of the frames bearing a bearing. Decision was made to carry all of the engine bearing loads in the outer casing and use it to keep the frames properly aligned. This provides lightweight by making maximum use of the material further away from the bearing axis. Carrying this a step further, a conical shaft, which is lighter for a given strength than a cylindrical shaft, is used to connect the three stage bearings with the compressor rotor.

### Structural Materials

Selection of the engine's structural materials and fabrication techniques centered mainly about handling the high stresses imposed by Mach 2 operation and providing the highest possible turbine inlet temperature for greatest thrust and lowest possible specific fuel consumption at cruise.

Steel was chosen as the primary material to be used in the hot parts of the engine.

Aluminum, which is finding favor with many engine manufacturers for its ease of use, was rejected because of its cut-off cross through General Electric had previously overcome its early problems in learning to fabricate the metal in the J79.

For Air Force models of the J79 only, the front frame and the forward compressor casing, the outer portion of the engine, are not made of steel. The front frame is a casting and the forward compressor casing is a mechanical forging made of a light alloy such as a magnesium-thorium alloy.

The remainder of the engine, the two rear frames and the rotor casing, are made of steel with stampings rolled together and designed to form integral units. The compressor rear frame and the turbine frame, which house the bearings and the shaft which carry the bearing loads and to the casing, are both made of stamped pieces in steel welded together.

The jet and welding techniques needed to produce these built-up assemblies in a volume basis represent a major accomplishment for the General Electric Aircraft Gas Turbine Division's production dept.

The Navy's crown of the J79 was steel throughout the engine because of the soft water corrosion problems with aluminum and the lighter aerobically. Built-up sheet steel construction is used as the front frame and the forward compressor casing. This reduces the engine's weight slightly but has the advantage of using the aluminum compressor inlet frame and the turbine inlet frame.

Other parts on the J79 that are made by welding stamped steel pieces together include most of the outer blades as the compressor. Compressor rotor blades, the turbine blades and some of the holder shrouds are forged in the usual manner. The turbine blades are attached to the wheels with simple tang for pins and the compressor blades are a single leg.

Compressor wheels are bolted together with tangs used between them to secure the blades in the wheels and keep them in place. This, in effect, gives the compressor a "crown" shaft. The crown shaft joining the three turbine wheels to the compressor is a

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solid piece of metal that is formed by a cold power spinning operation. This operation is similar to lead spinning on aluminum except that a machine takes a preformed flange of steel and shapes it around a rounded ring about 480,000 psi pressure.

The preformed flange that the machine begins with is 11 in. long. After a few minutes, the machine has formed it into the required overall shaft 31 in. long. Prior to the appearance of this machine from the Cincinnati Milling Machine Co., the shafts were made by machining a large flange so that about half of its original weight was lost in chips, and the fabrication time was twice as long.

### Phase II: Frost Frame

Frost frame of the Phase II engine will have eight shafts compared to four on the Phase I shaft members. Nine reasons for this is to improve the anti-seizing characteristics of the engine. Frost air is blown from the compressor through the frame frame shafts and the relief guide vanes to prevent the formation of ice and allow all-weather operation.

Compressor casing actually comes in three sections for the Air Force engines. A small casing is placed between the forward magnetism casing and the rear steel casing to allow for the different cases in three expansion qualities. On the Navy engines with all steel casings, only two sections are needed. All of the compressor casings are split lengthwise into two sections to allow easy inspection and maintenance.

The variable stroke blades on the compressor and the relief guide vanes are attached by two cylinders that are hydraulically operated on most of the early Phase I engines, but the latter have been changed and the leak pressure that provides the power for these cylinders before it is fed into the engine. This chamber is one of the hydraulic pumps from the engine accessory system.

Moving on from the compressor section, the combustion casing is in one piece on the Phase I models but has been split into two pieces for the Phase II production and adds a little weight but relieves a difficult inspection problem with the combustion. The case features chamber in all steel steel and is hybrid circular and circular design. It complex case situated in an internal section.

The turbine casing is split lengthwise on all models allowing easy inspection and maintenance of the turbine at the turbine section.

The shaftwork appears to be of conventional design and construction and it is constructed of the turbine frame and is otherwise unsupported. It employs a type of conventional flange locking that allows a removal of drag

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when the afterburner is not operating. In order to provide a high propulsive efficiency at supersonic speeds, the afterburner has to discharge the engine gas flow through a supersonic converging-diverging nozzle. The J79 makes use of cooling air flowing around the engine to help provide the nozzle effect.

The recirculator cooling air enters the afterburner near the air end and is bled so that it strikes the main gas stream nearly at right angles. Mixing of the two gas flows is negligible and the main gas stream is unaffected. To maximize the supersonic nozzle effect, the exhaust area of the afterburner is variable. It consists of interlocking steel fingers in a pivot arrangement. Power for this nozzle system is provided by hydraulic actuators actuated all at the engine oil system.

The variable area pivot arrangement which controls the mixing area of the afterburner is being modified on some of the newer designs to provide a lower base drag for the subsonic cruise condition (afterburner is not operating).

During subsonic cruise the required exhaust area is relatively small, and the petals are almost fully closed. To decrease the afterburner base drag in this condition, the length of the petals is being increased so that the effective friction ratio of the exposed nozzle is increased.

### Power Control System

Power control system for the J79 is tied together to the extent that the pilot has only one knob to operate. The main fuel system for the basic engine, the afterburner fuel system, the variable shaft control and the afterburner nozzle control are all integrated into one unit. Power for the control system is presently being derived mechanically with a direct drive. Information needed by the system to operate properly is believed to be limited to turbine discharge temperature, compressor inlet temperature and engine rpm.

The engine is mounted in an aircraft by the front compressor frame and the turbine frame. Side and vertical loads are taken out through the compressor front frame. Two suspension points are available there, and two of them must be used.

Thrust loads are taken out through the turbine frame which has three universal fittings for thrust pins. The B-58 uses one all three and the F-104 two. All side loads are taken out through four tangential mounting lugs on the turbine frame.

Engine lubricating system is self contained with the except pressure provided from the compressor and the cooling from the duct before it is bled.

## U.S. Army Interested In Voltigeur SE-116

Paris—Interest shown by the U.S. Army in Sud Aviation's new turboprop multi-purpose aircraft, the SE-116 Voltigeur, has led to license talks between the French company and Rockwell Aircraft Corp.

Voltigeur first flew in early June (AW Aug. 10, p. 65).

Initial flight tests are being carried out with two Curtiss-Wright R1500 powerplants. These engines will be replaced by two Turbomeca Baston turbo prop engines producing 740 each.

The 116 model is a three-seater which cruises at about 280 mph. Landing gear is tricycle type.

United States Army interest reportedly centers on an enlarged version of the Voltigeur which carries the designation 117. The main difference between this aircraft and the present Voltigeur is an enlarged fuselage which will permit carrying of five passengers in addition to the crew. Prototype of the 117 is presently under construction and the company expects to fly the aircraft by the end of this year.

The 117 Voltigeur would be a multi-mission aircraft. Besides liaison chores, the aircraft would fly observation missions and be equipped to make ground strikes once a target is established. The 117 reportedly would fly, some seven hours. Gloster takes up a prime position in a Pheasant aircraft section. Photos of the present Voltigeur include an interchangeable pack, its under the fuselage which provides a wide continuous instrument or photo exposure.

Voltigeur was developed by Sud Aviation at the request of the French government. Idea behind the project is to provide the military with a multi-purpose aircraft which can be used in local conflicts where there is opposition or potentially nonexistent.

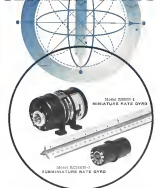
Voltigeur is one of four French projects designed to fit this concept. At present it seems that Sud Aviation's entry captures the most favor. Other projects are Messier-Schneider's Eclair and SERA's 1100. The Eclair has fixed gear and is powered by one Baston turboprop. The SERA aircraft presently is powered by two Pratt & Whitney R1140 engines.

First SERA prototype carried its aircraft, killing the test crew. Second prototype is under construction despite reports that the Voltigeur is the winning entry.

Later on to competition, AVATRAC West has learned, is the Dassault 115 Flammant. Capable of carrying eight passengers, the Dassault aircraft is powered by two Baston turboprops. The aircraft is expected to make its first flight in December.

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\*Illustrations show the 10 of the 31 different Cornelius compressors available.

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Large portion of Jentel's product line is in components for high temperature pneumatic systems which handle oxidized pressurization, fuel tank pressurization and wing and tail actuating in jet aircraft. These systems, which bleed high temperature air from around turbine engines, include in components pressure regulators, shut-off valves, flow limiting valves, inert gas generators, evaporative coolers, heat exchangers, air coolers, duct couplings and supports.

Quick-disconnect duct couplings are made of titanium and produced by a hot forming technique. Flanges, however made of stainless steel, are used to provide a light, strong coupling for pneumatic ducts at temperatures up to 1500° when welded to ducting.

Master compressors now comprise only a small portion of the company's sales total. Pneumatic controls for missiles, including the Regulus II and Talos, are being sold by Jentel and the division appears to be intensifying its efforts in this area.

Heavy lot ground support of both military and aircraft are produced by the division. Combination heaters are used to heat missile fuel and removed wet loadings.

Division also has one hot fuel pump worth for starting, preheating engines under extreme cold-weather conditions.

Division's own plant, occupying 130,000 sq. ft., is designed to provide extensive facilities for both expanding and competitive testing. Jentel, specialising in precision metal forming operations, has built only through extensive testing run at its end the weight position of expanding and expanding metal by hydrostatic water storage.

High altitude test chamber in operation at the new plant is capable of simulating altitude conditions at 101,000 ft. cylindrical chamber, measuring 20 ft. long by 6 ft. in diameter, has a test specimen size capable of simulating a pressure altitude of 50,000 ft. and a temperature of -65°, while testing a combination heater delivering 1,000,000 Btu per hour. Altitude loop will be used to test heat exchangers, combination heaters, pneumatic controls and duct couplings. Prior to the installation of the chamber, components had to be tested at windtunnels scattered throughout the country.

Plant is designed to facilitate introduction of design modifications at any point during production, Jentel says. Production lines are streamlined and customer and production processes are combined for each particular job. This system, the company says, provides maximum flexibility in the manufacturing of pneumatic components in relatively small quantities.

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# HAYSTACK



In it is a needle, mined and capable of leeching needles. Your Navy searches the oceanic haystack with Grumman S2F Trackers. These carrier-based airplanes can detect an invisible submarine by its subtle deflection of the earth's magnetic field. Their sophisticated equipment can spot a sub's floating peak with attack radar. They can lay electronic ears on the surface to hear a sub move bottoms below. Once contact is made, Grumman Trackers attack with weapons that silently sink and strike their prey stalking along the bottom.

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GRUMMAN AIRCRAFT ENGINEERING CORPORATION  
Bethpage • Long Island • New York

AVIATION WEEK, August 15, 1958



Clean room includes ultra-violet light inspection (left) and pressure test rig (right).

## Ultrasonic Device Cleans Valves

Ultrasonic cleaning and ultra-violet light inspection are two methods used to insure cleanliness of valves for missile ground support equipment. Cleaning of the internal components—including threaded ports and valve seats—is imperative with valve used with gaseous and liquid oxygen and nitrogen which may pass through minute control surfaces. Valves must be free of dust and lint particles, and more particularly, free of all oil grease.

Supreme Valve and Fitting Co., Pittsburgh, Pa., maker of manual control valves for Douglas Aircraft, carries out cleaning, assembly and inspection of its

valves under controlled atmosphere conditions. Exact applications of the valves is classified, however, valve in photographic appears to be a liquid oxygen valve, which would be used on ground support equipment for the Douglas Thor intermediate range ballistic missile.

Valve components, brought into the clean room are first cleaned ultrasonically.

Parts are mounted in the cleaning tank and subjected to 18,000 cycles per second vibrations induced in the cleaning fluid by a crystal-controlled transducer. Vibration loosens the forces of adhesion binding small particles in the valve components. Pressure is said to remove 90% of the dirt particles down to one micron in size.

### Ultra-Violet Inspection

After ultrasonic cleaning operation, components are taken into an exact clean room and inspected under ultra-violet light, which will reveal even trace quantities of hydrocarbons. Operation wearing rubber gloves they assemble the valves and place them on special test fixtures. Valves to be pressure tested are mounted on a special rig and permeated with nitrogen which is stored outside the clean room and piped to the fixture through a 50 micron filter. Approved valves are immediately packaged in heat sealed polyethylene bags.

Valves remain in the plastic bags until just prior to installation on missile ground support equipment. Clean room, built in cooperation with Douglas, is typical in trend to others in the aviation industry. Room consists of two



Valves are packaged in heat sealed polyethylene bags for shipment.

# MINNIE CONNECTORS

pass tough, new  
ALTITUDE-MOISTURE  
RESISTANCE TEST  
salt water immersion,  
65,000 feet altitude

AMPHENOL



Designers and manufacturers of aircraft and missiles, as well as the military, have long recognized the need for a connector altitude-moisture test which would accurately simulate actual performance conditions. Such a test has been developed by manufacturers and the military and applied as standard procedure on the 67 Series AMPHENOL connectors in the AMPHENOL Laboratories. In compliance of the following:

Following the altitude-moisture resistance test, insulation resistance measurements (in megohms) on production AMPHENOL "E" connectors were as follows:

Days	Insulation Resistance (in megohms) on production AMPHENOL "E" connectors were as follows:	Days
1	1000	200
2	1000	200
3	1000	200
4	1000	200
5	1000	200
6	1000	200
7	1000	200
8	1000	200
9	1000	200
10	1000	200
11	1000	200
12	1000	200
13	1000	200
14	1000	200
15	1000	200
16	1000	200
17	1000	200
18	1000	200
19	1000	200
20	1000	200

AMPHENOL AMPHENOL "E" connectors not only meet but exceed the requirements of this tough new test. 300 megohms is the minimum insulation resistance required by MIL-C-5033C after moisture. AMPHENOL minimum insulation resistance after immersion and altitude cycling is 1000 megohms.

## 67 Series AMPHENOL "E" Connectors

DESCRIPTION: Miniature, multi-contact electrical connectors of the quick-disconnect bayonet lock type. Available as Plug, Cable and Panel Receptacles, and Single Hole Mounting Receptacles. Shell design classes include:

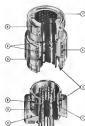
- class A—Environmentally resistant—industrial wire seal
- class B—For potting
- class W—Environmentally sealed
- class F—For potted cable
- class C—Standard cable clamp

There are five shell sizes, and 12 insert arrangements—ranging from 3 contacts in the smallest to 48 contacts in the largest.

PART NUMBERING: Descriptive part numbering of AMPHENOL connectors follows that used with ASG (ASG) connectors.

NOMINAL CURRENT RATING: #20 contact is rated at 7.5 amperes and #16 contact at 12.5 amperes.

OPERATING TEMPERATURE: —67°F. (–50°C.) to +257°F. (+125°C.).



HOODED SOCKET CONTACTS



Both #16 and #20 socket (female) contacts of AMPHENOL AMPHENOL connectors are resistant to test panel damage. The entering end of the socket has a single-point lead that excludes the entrance of a pin 0.005" larger than the diameter of the mating pin. AMPHENOL Specification 300-13, 200, paragraph 4.3.14, gives this test to be used to determine resistance to test panel damage.

A test of hooded socket (female) contacts of AMPHENOL AMPHENOL connectors is required to test panel damage. The entering end of the socket has a single-point lead that excludes the entrance of a pin 0.005" larger than the diameter of the mating pin. AMPHENOL Specification 300-13, 200, paragraph 4.3.14, gives this test to be used to determine resistance to test panel damage.

After withdrawal of the female at the completion of the above procedure, the force needed to engage or disengage the socket contact shall not exceed the following values:

Contact	Max. Force Down	Max. Force Up
#20	10	2
#16	20	2

## FEATURES OF AMPHENOL AMPHENOL CONNECTORS

- Environmentally sealed with unitized lock and grommet. (Also available with provision for potting.) Grommet and (type "E") means altitude-moisture resistance requirements. Either grommet and/or potting seal meets moisture resistance requirements of MIL-C-5033C, Paragraph 4.3.13.
- Spring-loaded coupling ring provides a positive locking action in the bayonet slot, and a constant compressing force which eliminates the effects of resilient face and compression set.
- Stainless steel bayonet slots and pins reduce wear and friction characteristics and eliminate wear associated with "hard-core" and similar surface treatments of softer base metals. The three pin bayonet coupling minimizes the rocking action of the mated plug and receptacle.
- Plated (in-line) angle of bayonet slots and area mating face requirements.
- Hooded contacts resist test panel damage as defined in Paragraph 4.3.14 of AMPHENOL Specification 300-13, 200.
- Unitized grommet seal, clamp and grommet forms a single unit for ease of assembly and maintenance.
- Face and gasket with individual barriers to isolate each contact.
- Hard insulating dielectric (glass reinforced fiber seal) permanently retains contacts with no possibility of contact being pushed out of the insert.
- A visual full engagement indicator is included in the design to insure the user that he has fully engaged the connector. The indicator is an orange line around the receptacle shell.
- Insulation resistance of "E" type following altitude-moisture resistance test is a minimum 1000 megohms MIL-C-5033C minimum following type "E" test is 100 megohms.
- When used mated sealed connectors, no derating for altitude is necessary at 75,000 feet.
- Test voltage 3,000 volts 30/30 75,000 feet on sealed connectors.
- Vibration per Method 204 of MIL-Std-202A, 10 to 2,000 cps at 20 g.
- Temperature cycling range per MIL-C-5033C, Paragraph 4.3.2 according to 257°F maximum and –67°F, minimum.

## VOLTAGE RATING

	Rating	Maximum Working Voltage (in volts)	Maximum Working Voltage (in volts)	Maximum Working Voltage (in volts)
Test level	A	250	250	250
Test level	B	500	500	500
Test level	C	1,000	1,000	1,000
Test level	D	2,000	2,000	2,000
Test level	E	3,000	3,000	3,000
Test level	F	5,000	5,000	5,000
Test level	G	10,000	10,000	10,000
Test level	H	20,000	20,000	20,000
Test level	I	30,000	30,000	30,000

The "E"-type construction of AMPHENOL 67 Series AMPHENOL connectors was originally designed to meet the moisture resistance requirements of MIL-C-5033C, Paragraph 4.3.13. Since the development of the new test for more stringent altitude-moisture test, AMPHENOL AMPHENOL design has been modified and all AMPHENOL AMPHENOL "E"-type connectors pass this test.

Send for your copy of the AMPHENOL catalog to complete information.  
AMPHENOL ELECTRONICS CORPORATION  
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CONFORM TRENTWELD TUBING is specified in a wide range of grades including stainless steels, titanium, Inconel, Monel, 414 9-14 — and in 12 basic tubing shapes and sizes in wall thickness, outside, inside, outside, as well as, seamless, welded, and flared tubing. Trentweld of Trenton, N.J.

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Furthermore, Trent checks the quality of its pipe and tubing with a series of strict tests. All strips going into the welder is checked for width and tolerance. Samples of each lot are tensile tested. Fatigue tests — flattening, reverse bend, flare and flange, coil and pressure — are conducted. Pipe and tubing for corrosive applications get rigid corrosion tests. And, if necessary, "temple-

ment" X-ray inspections, exclusive with Trent, can be made on any lot.

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Just samples for the above pipe being tested — showing uniformity in physical, chemical, mechanical properties, etc. Trentweld of Trenton, N.J.

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at C. Wallace, General Manager, Bendix-Pacific, Southern California Division, 401 Main Street, Los Angeles 12, California. I am interested in: (1) General Engineering (2) Mechanical Engineering (3) Aeronautical Engineering (4) Electrical Engineering (5) Chemical Engineering (6) Other (Specify): \_\_\_\_\_

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chambers, one inside the other. Outer chamber is used to perform the ultra-sonic cleaning operation and the inner chamber is used for final assembly, testing and packaging.

Two this air conditioning unit is employed to clean the air in the rooms in size of 10 hours per hour. An electric filter capable of filtering dust particles down to one micrometer and thousands of a micron also is used. Vacuum type venting system removes fumes from the technicians from cleaning tanks by maintaining an under-ventilated atmosphere. One cubic foot of air per minute enters an exhaust fan of air when the door is opened.

## Navy Drone Launched By Compressed Air

Simple, compact and light weight are said to be the advantages of Navy's elastic cylinder type drone launcher developed by the Naval Air Engineering Center, Philadelphia, Pa.

Launcher's design is such that the drone is attached, centered in the elastic cylinder which is flattened length of standard inch diameter heavy duty hose. To fire the drone, crew presses on a button on the hose forcing the drone to exit the opening of the launching tank.

By increasing the pressure in the hose during the fire, the drone is accelerated and velocity is kept the drone flying until it reaches its target. Carriage only undergoes drop of about 100 feet, but has been completed.

Navy says production model of its elastic cylinder drone will be one half the weight and 90% less expensive, to operate than piston and jet-powered rocket drone launchers.

Production models of the new launcher also will differ from the old one in that the tank will be an integral part of the launcher. At launch time it will be held with flanges at both ends so that multiple of drones can be attached together to launch several drones.

Tests indicate that one section would be sufficient to launch a 175 lb drone with initial velocity of 85 mph using 500 psi air pressure.

Three sections in combination would launch a 1,000 lb drone at 200 mph, said to be the maximum requirement for drone launching equipment.

Navy is considering further applications of the elastic cylinder launcher, including use of a larger launcher and liquid gas propellant to accelerate a vehicle to near space speed. The launcher would be a substitute for steel tube launcher and piston and jet launchers to carry a considerable number of drones.

## LIBRASCOPE analog digital converters shift to digital encoders



Features of Librascope Shift Point shift to digital converters include: modular design, built-in digital-to-analog and analog-to-digital converters, built-in digital-to-analog and analog-to-digital converters, built-in digital-to-analog and analog-to-digital converters. They are designed for high speed of data from a shift point in a large number of data logging systems. Special computer time sharing in shift to digital converters. Inbuilt time sharing computer. Inbuilt time sharing computer.

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small trucks and carts, and on small boats to service advanced jet airplanes. For complete information write to Dept. P-52, Solar Aircraft Company, San Diego 92, California. We'll include a free, illustrated brochure.



or incorporated with the exception of one subsonic variable valve, in inlet stage for high impedance matching purposes to existing vintage systems. It has the following characteristics: thrust, 11 lb; air flow velocity, 15 in./sec.; stalled thrust, 150 lb.; repeatability,  $\pm$  4%  $\pm$  .007 in.; total weight under 1 lb. Price of actuator, amplifier combination is from \$729 to \$768, depending on quantity.

Advanced Research Associates, Inc., 4850 Howard Ave., Kensington, Md.

### Water Injection Pump

Jet engine water injection pump is designed to run dry after supplying thrust augmentation as takeoff.

Model 7751 pump, in use on J57 engines, delivers water at 90 gal. per min., 400 psi, operating at 1,500 rpm. After takeoff, pump is disarmed and cooling jets continue to operate for duration of flight. Operational temperatures vary from -60 to 740 F. Unit incorporates a liquid expeller, vapor element and a relief valve. Diaphragm prevents damage to prevent icing. Pump weight is 18 lb.

Chandler-Evans, West Hartford 1, Conn.

## WHAT'S NEW

### Publications Received

**Richard The Spectra-In 1**, J. Kasper, ed., Public Affairs Press, Washington, D.C. \$6.90, 195P.

A survey of Soviet submarines from its pioneering days, up to the launching of its first artificial earth satellite.

**Missile Engineering Handbook**—by Cassius M. Hall, Capt. USN (Ret.)—Pub. D. Van Nostrand Co., Inc. 120 Alexander Street, Princeton, N.J. \$12.50, 839P.

This book is the fourth volume in the series entitled, "Principles of Guided Missile Design." This series is intended to give well-grounded material on the technology of guided missiles and space flight.

**Avionics in Stability and Control**—published by J. B. Riet Co., Inc., for Wright Air Development Center, U.S. Air Force, 5800, 795 pp. (Replaces P8 151027 from Office of Federal Services, U.S. Department of Commerce, Washington 25, D.C.).

This volume presents technical material which enables a practicing designer to understand various aerodynamic phenomena and to apply methods for investigating aerodynamic effects in equations of motion. Texts appear also given for solution of the equations. Material is presented by "example" rather than by a manual

approach, since many of the techniques of analysis are still being developed.

**Optimal Gasdynamics**—by Dean Politz, published by Longmans, London, 11110 Buchanan Drive, Wichita 4, Kan. 368 pp.

History of the Army's light planes is covered in Korea from aggression to armistice, 1910-1953.

**Digital Computer Components** and **Control**—by K. K. Richards, published by D. Van Nostrand Co., Inc., 120 Alexander Street, Princeton, N.J. \$10.75, 325 pp.

This book provides a ready source of

basic engineering approaches related to digital techniques, carefully organized for practical use. Logical functions and digital design, basic operations to be performed by a digital computer, are discussed at length.

**Via Nostrand's Scientific Encyclopedia**—published by D. Van Nostrand Co., Inc., 120 Alexander Street, Princeton, N.J. \$16.00, 1,508 pp.

This work covers the principles and practical applications of the past, present and biological sciences, mathematics, engineering and medical technology in a book both handsomely and cost-effectively for convenience.



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NERVE CENTER OF THE NEW INDUSTRIAL ERA

## WHO'S WHERE

(Continued from page 35)

### Changes

**Joe Y. Bowman**, project manager for the USAF Maintenance Center's guidance and control system, Autonetics, a division of North American Aviation, Inc., Downey, Calif.

**Richard F. Fournier**, director sales and service, and **Harold D. Gilbert**, manager contracts, Raytheon Missile Division of Theodora Chemical Corp., Danville, N.Y.

**David E. Albrecht**, program manager responsible for advanced large scale engine control, Rockwell, a division of North American Aviation, Inc., Canoga Park, Calif.

**Walter R. Woodward**, director of engineering, Dynamics Corp., Burlington, Mass.

**Jack G. Anderson**, director military sales, Lockheed Laboratories Division, Hoffman Electronics Corp., Los Angeles, Calif., and **Tom C. Clark**, marketing manager for the division's recently established Electro-Mechanical Department.

**Frank Belencki**, chief engineer, Wyle to Autonics Corp., North Hollywood, Calif.

**E. L. Labin**, assistant representative in charge of all Thermo-Program operations, Aerojet-General's recently established field service office at Douglas Aircraft Co., Tulsa, Okla.

**Doyle W. Korfman**, manager, Timbren and Vantage Missile Products Division, Grinnell Steel Company of America, Pittsburgh, Pa.

**Frank W. McCabe**, special assistant to the manager, Rockwell Corp., a Division of Autonics, Boulder, Colo. Also, **A. E. Clark**, manager engineering and sales.

**Robert E. Korfman**, manager finance, market planning and development, Penn Products and Products Division, Raytheon Corp., Bedford, Mass. Also, **R. E. Harkin, Jr.**, general manager, and **Alvin F. Feltz**, operations manager, Penn Products, newly established Western Branch, Los Angeles, Calif.

**Howard D. Wessman**, manager of site test procedures and analysis testing, operations, Alvin S. De Mont Laboratories, Inc., Cambridge, Mass.

**Carlton E. Robinson**, customer service manager, Autonetics Division, Autonics, Inc., Hawthorne, Calif.

**Michael Baker**, manager manufacturing and engineering, Semiconductors Division, Schottky Electric Products, Inc., New York, N.Y. Also, **Dr. Samuel Myerling**, chief engineer, and **Theodore E. Kowalski**, sales engineering services.

**John F. Harpiss**, manager Dayton, Ohio, office, Good Rapids Division, Raytheon, Good Rapids, Mich.

**Vernon G. Edwards**, special representative sales and service, Dallas Autonics, Inc., Dallas, Tex.

**Wynne L. Harty**, director of marketing, Rockwell Corp., Whittier, Calif.

**D. E. Schaffer**, Eastern sales manager, and **Charles H. F. Schaffer**, comptroller, Hydrex Corp., North Hollywood, Calif. Also, **E. G. Gage**, district sales manager, General Aircraft Co., Wichita, Kan.

## ENGINEER OPPORTUNITIES AT RAYTHEON



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## 1008

## LETTERS

### Thermal Soaring

As a confirmed reader of *ANASTAS WOOD*, I have long wondered how something so interesting as thermal scoring could be overlooked. Robert Stoufford's article on *Schwann Anzette* (*AW* [July 14, p. 58]) marks the plants of the scoring plant he got on out of the label stick, using and rig plant into scoring machines. An open invitation is extended to Mr. Stoufford to try out the Florida brand of thermal that

John J. FARRAR  
Capt., F.A.L.  
Comd. Co. Med. Flt.

## Goose Engine

You stand AVIATION WEEK on pulled a horse. In the picture on the far right, a plane Division "Goose" Mustang and across jumping information on your face 13 years in 271. It is stated that the engine used in the Mustang III. Actually, the engine is installed in the ported aircraft in the Air Force's Mustang Viper. Mustang which Costa Wright supplied Fitchell after a lawsuit was filed with Mustang Viper Mustang Ltd. A number of Viper 3 engines have been and are still being used successfully in the development of the "Goose" Mustang. To the best of our knowledge, the FBI has not or not actually prevented one of these aircraft in flight.

Knowing how you grade yourself on facts, I thought you'd like to be set straight so you could use a correction.

Raymond S. Goss,  
Director of Public Relations  
Carter-Winget Corp.  
Windsor, N. I.

### Courtesy by Mail

On looking through your issue of May 18, I was very interested to notice an advertisement by a well-known helicopter news feature, in which one was invited to drop them a line, to which was added "Please return enclosed."

For the first time it has been a source of amazement to me why firms in the United States and I do not mean only in the private industry, spend thousands of dollars individually on advertising and on public relations, yet when it comes to the one important item that really represents a company and their method to look at itself, in the case even be, in dealing with this, namely a firm's correspondence to me, can either ignore the allegation or take their own next step for the eventual court on, of an acknowledgment.

I have found this particularly to be the case when seeking employment in aviation as I have been left searching, and in this connection it is to be passed out spent releases from any number of good will that may be concerned. But the person who may be seeking employment today—well, who knows—tomorrow may be in possession of another position to do business with those shortchanged enough to ignore his previous work.

*Arctian* Week welcomes the opinions of its readers on the issues raised in the magazine's editorial columns. Address letters to the Editor, *Arctian* Week, 535 W. 42nd St., New York 36, N. Y. Try to keep letters under 500 words and give a genuine identification. We will not print anonymous letters, but names of writers will be withheld on request.

In a country in Europe, in which I have been in the rubber business for some time it is the usual practice that if a letter cannot be taken care of right away the use of a number of reasons is resorted to and excuses are piled upon excuses to avoid the business concerned is being taken care of and also makes for a feeling of good will towards the firm concerned, as, to it is usually put here, shown that "they are on the ball."

Some time ago I was heading the sales of subscriptions, etc., for an amateur journal, and in addition to an immediate acknowledgment of all checks for subscription fees, etc., I sent a courteous letter thanking them for business. At the same time, as compliments, such as papers not received, etc., were immediately acknowledged the matter was expedited, and satisfaction made. The result was that I not only did not lose a single subscriber, but sold further subscriptions in existing subscribers in addition to good will sales from them, and later advertising sales on these same people.

There are such simple points that it is incredible to me that they should be ignored, and how small is the cost compared to money that is often wasted on other things.

It is only necessary for me to add, as I gave thousands that have done before me, that I consider *AVIATION* *Worth* to be first class in every way, and not only a luxury of reading matter but an absolute necessity if one is to keep "on top" with world aviation news.

ROBERT G. DE CROMBIE  
New York, N. Y.

### Control Problem

Have just finished your issue of Feb. 21 and find once again a letter "Flying Coasts Blind" (p. 102) expressing the principle of "let's not be seen" as the answer to making inflation. I continue to wonder why you print such blatant Hanning all sides of an issue, a definitely desirable. However, I humbly submit that anyone who offers an untested insight publicly in the manner you speak of, is completely ignorant of the problems involved in measuring high speed winds. (I don't recall any comments by pilots of such aircraft to the contrary do you?)

The frequent retention of the no dual income but rising point of view leads not only to confusion. Viability is a desirable thing and as an industry representative I have urged the Civil Aeronautics Board to require even those cockpit visibility systems and systems, however, for a variety of reasons. Increased cockpit visibility will unquestionably make the smallest contribution of any

citizen leaders to improve our safety as regional railroads. We are dealing with a problem of traffic control and nothing else.

So please—let your law apparatus for stupidity to clear the air. Let us see the problem for what it is and work together to solve it.

Such emotional concepts as the "rights" of this group or that simply do not fit into such important (rational) deliberations.

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 Atlantic City  
 New York, N. Y.

## From Ground Up

In reply to Mr Morgan's letter (WM, July 7, p. 110), I would certainly go along with the policy of aircraft flag IFR flight plans, but what about the thousands of civilian (private) aircraft and pilots who could not qualify for IFR flight?

In my letter (AW June 23, p. 94), my suggestions on paint and lights were limited to supplemental aids, not a cure to the mid-air collision problem.

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### Collision Solution

In his letter "IFR Filing" (AOW July 7, p. 100) Copilot Martin wonders if fact-

"1. There is no record of two aircraft having Mod and flow IFR running into each other. 2. In each case of a similar situation one or both aircraft using VFR, thus making the accident possible."

If Mr. Marton will write to the C&M they might have a spare copy of Aerial Accident File #1417-42 dated June 21, 1945.

It officially records a mid-air collision in the narrowest. Both aircraft had Med. Power and complied with IFR flight plans.

Mr. Moran has offered a simple solution for the mid-air collision problem, backed up by his facts: The assembly of voice solutions is for everyone to be FRK (above this, that, or some other dither— it's hard to get down where the FRK should start).

In the moments after pilots wonder how long an overcrowded traffic control system can continue without another (other) aircraft capping up— to say nothing of the traffic that just doesn't get a chance to go FRK because of delay.

It seems to me that the sole and ultimate will be an effective anti-collapse electronic warning device in the cockpit. Until now a perished pilot will have to keep looking out the window regardless of what kind of flight plan they may file.

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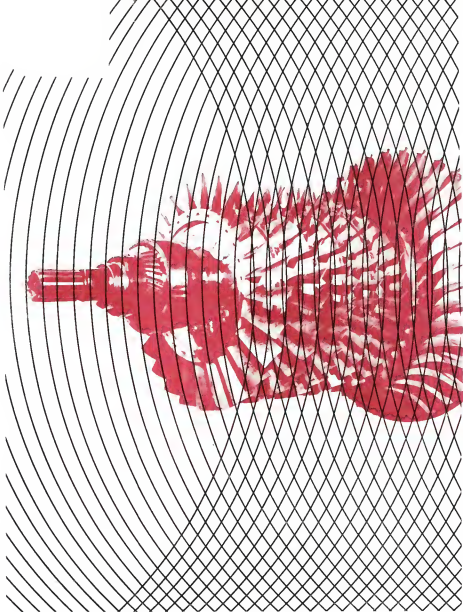
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